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Geosynchronous SAR Technology – Learning from Hydroterra (Back to SAR basics)

Dr Steve Hobbs

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Lessons learned during the Hydroterra Phase 0 Study

- 1. Relearning SAR basics: airborne not LEO SAR
- 2. Achieving useful SNR
- 3. Geosynchronous orbit
- 4. Polarisation options for GeoSAR
- 5. Radar hardware challenges
- 6. Operations, an Earth Science Observatory

Discussion and Conclusions





More like airborne SAR than LEO SAR

Can image area of interest whenever required

- Squint viewing and swept spotlight mode in general
- Integrate for long enough to build synthetic aperture needed for the desired spatial resolution (e.g. ~1 min to achieve 1 km resolution in C-band)
- "Zero Doppler" thinking from LEO SAR is not particularly helpful

Incidence angles range from ~20° up to ~65°

- Blind to sub-satellite area (rainforest, ocean are unstable anyway)
- Need high incidence for high latitudes (and LEO is good here)
- Regions with high incidence angles are usually most challenging because signals tend to be weaker

Very good low latitude coverage – complements LEO SAR



Earth disk from 20°E in GEO: shaded torus for 20° < incidence < 65°



Aperture, power, integration time vs spatial – temporal resolution and coverage

From GEO it is a challenge to achieve a useful SNR

- Large aperture for high EIRP and to collect signal power
- High transmit power (few x 100 W RF; and low prf)
- Long integration time (= high integration gain)
- Coarse spatial resolution (few km easy to ~50 m more challenging)
- Low relative orbit speed (m s⁻¹ vs km s⁻¹) helps us

Data: backscatter and InSAR products

- Multi-looking needed for most products
- Phase-based measurements are most robust easier to retrieve from noise than σ^0

Interference / noise

• RFI and clutter (surface, atmosphere) both need careful management – and some surfaces cannot be imaged (e.g. water, wind-blown forest)

 $SNR = \frac{\overline{P_t}L_x A^2 \eta_t \eta_a \sigma^0 \cos \theta}{8\pi v_{az} \lambda R^3 F_n kT_s}$ enging) $Lose factor ~40^3$ wrt LEO SAR



Wide choice of orbits; Significant perturbations; Major influence on mission design

Geosynchronous orbit

- Inclination controls N-S motion
- Eccentricity controls E-W motion useful to synthesis aperture
- Wide range of ground tracks is possible by relative phasing of inclination and eccentricity *Ignoring short-term* oscillations

Perturbations

- Luni-solar gravitational perturbations dominate (~50 m s⁻¹ yr⁻¹)
- Expensive to fight directly, but must be managed

Orbit determines relative velocity

- Orbit has a major influence on mission design:
 - Dwell time, importance of atmospheric perturbations, azimuth resolution, geographical coverage, payload requirements, size of image stack for interferometry, ...



Example orbit ground track during 1 month with no control; InSAR requires orbit repeat within ~0.03° (~20 km) in latitude (baseline decorrelation, height ambiguity, etc.)



Compact or Quad-Polar are most attractive

Using the same GEO location for all target areas means that a rotation is needed from satellite axes (nominally East and North) to the local H, V components

For a single linear polarisation at the satellite this results in a position-dependent slant linear polarisation at most locations

• Not satisfactory (and cross-polar term is too weak)

Compact-pol: transmit L or R, receive E, N \rightarrow H, V

- Good signals in both channels
- "80% of the information" wrt quad-pol

Quad (full) polar

• Full polarimetric information, but need twice the power since transmitting two polarisations alternately and cross-polar terms are noisy (poor SNR typically)



A position-dependent rotation is needed to convert <u>satellite</u> polarisation components (N, E) to local H and V at the <u>target</u>



Lightweight deployable antenna; RF components for high power / low prf

The main technology drivers for GEO radar are:

Antenna – large aperture area (circular is most suitable) – an obvious need

- Large, lightweight deployable reflector
- Low-cost (and European)

High RF power (C-band) – also obvious

 Components suitable for high power and low prf (~ few x 100 Hz) – "devil in the detail"

Getting to GEO

- Well-understood, but not cheap
- Electric orbit-raising is widely used leaves spare power for SAR



LSS 5 m diameter deployable antenna deployed at ESTEC (2016)





Complex operations planning; Software-defined; Observatory concept

User requirements – combine several types of mission

(a) Routine observations, (b) "Campaign" mode,
(c) Rapid response for emergencies / opportunities

Multiple constraints – AI management task?

 Orbit maintenance, InSAR coherence, solar / sidereal periods, orbit turning points, sufficient #looks, ...

Fully software-defined: no extra consumable

Earth science observatory

• Point to target of interest as and when needed; users could book time as for an astronomical telescope



Observations use the same orbit portion (<u>sidereal</u> period) to obtain InSAR products: hence drift of 2 hr per month in operations schedule (in <u>solar time</u>) **Discussion and Conclusions**

Discussion

Cranfield University

- ESA Earth Explorer Phase 0 study advanced our understanding significantly of both the science and the implementation
- Cost was Hydroterra's primary challenge $\sim \in 300M$ (implied; but no published costs)
- Direct launch to GTO / GEO is preferred
 - LEO + EOR is possible, but gives minimal cost saving + marginal mass budget
- Science / technology: no show-stoppers, but some clear challenges Even with Phase O scrutiny

Conclusions

- Team remain convinced of the science potential and technical innovation of GEO SAR
- Priorities:
 - Hardware development and de-risk (antenna, RF components)
 - E2E simulation incl. full product retrievals in realistic scenarios
 - Science maturation, incl. polarimetry options and benefits for Africa



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Thank you

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s.e.hobbs@cranfield.ac.uk