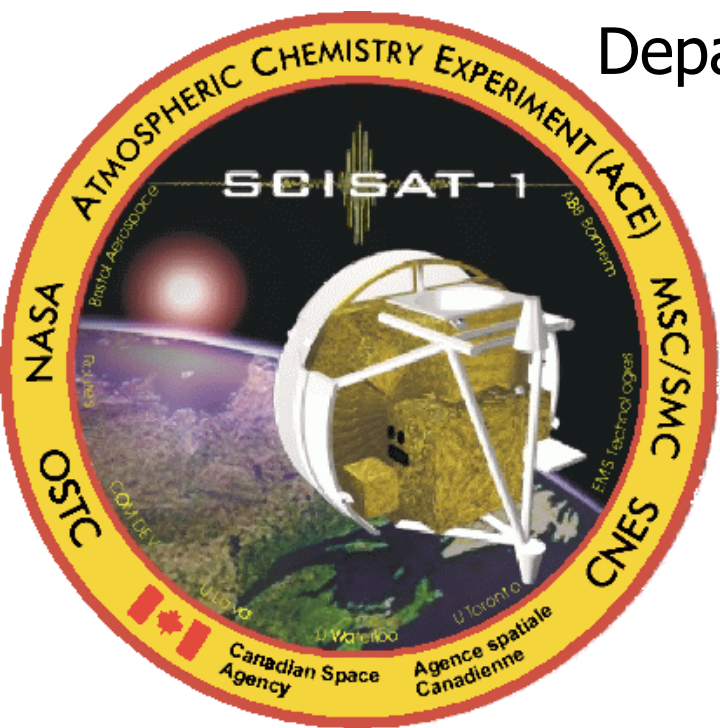




# Small Science Instruments: The Atmospheric Chemistry Experiment (ACE)

Peter Bernath

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Heslington, York, UK



THE UNIVERSITY *of York*





**Inspiration:  
ATMOS on  
Space Shuttle;  
Fourier  
transform  
infrared  
spectrometer  
(FTS) working  
in solar  
occultation;  
flights in  
1985, 1992,  
1993, 1994  
(Farmer at  
conferences)**

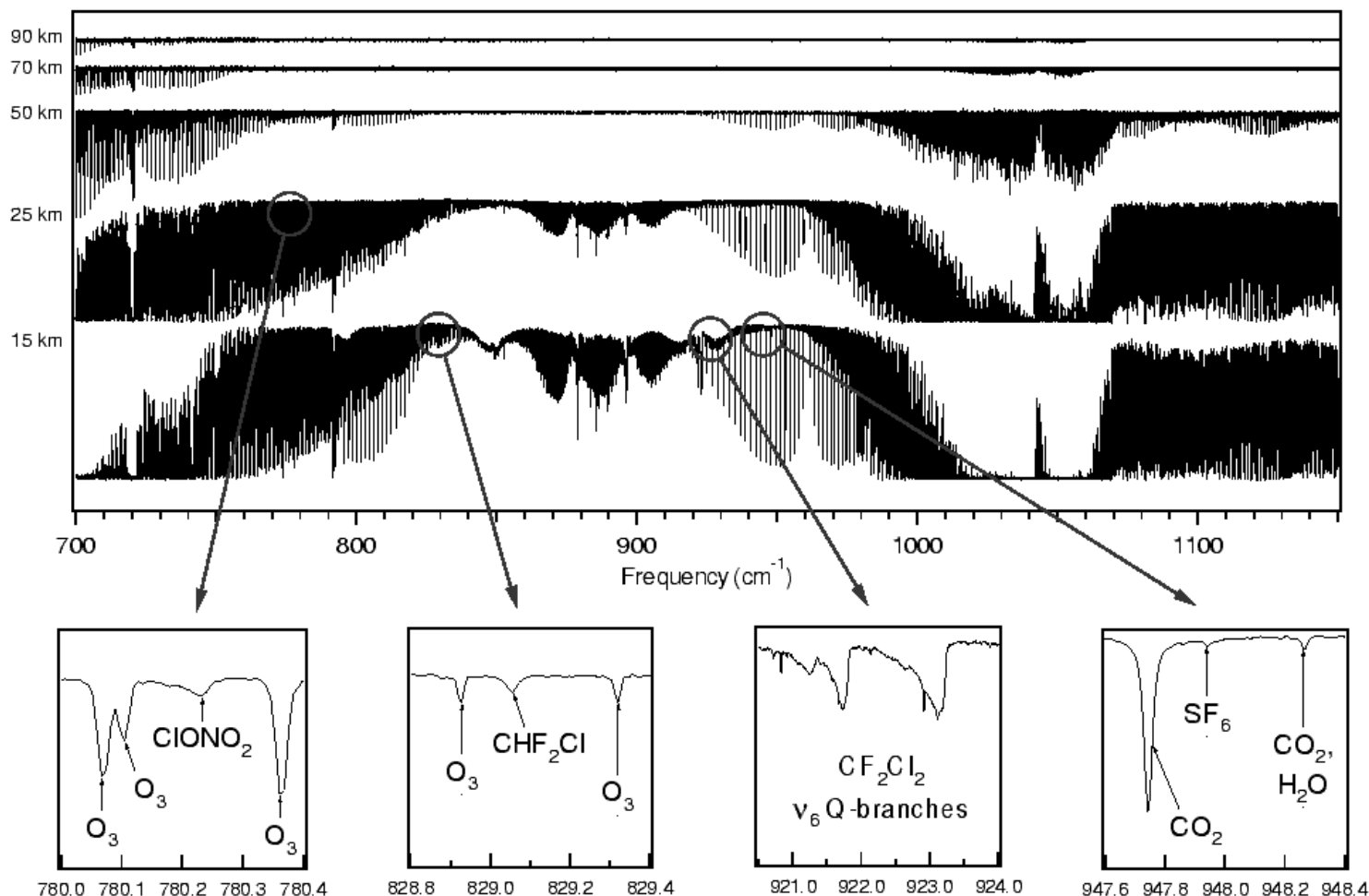




# ATMOS IR FTS Spectra

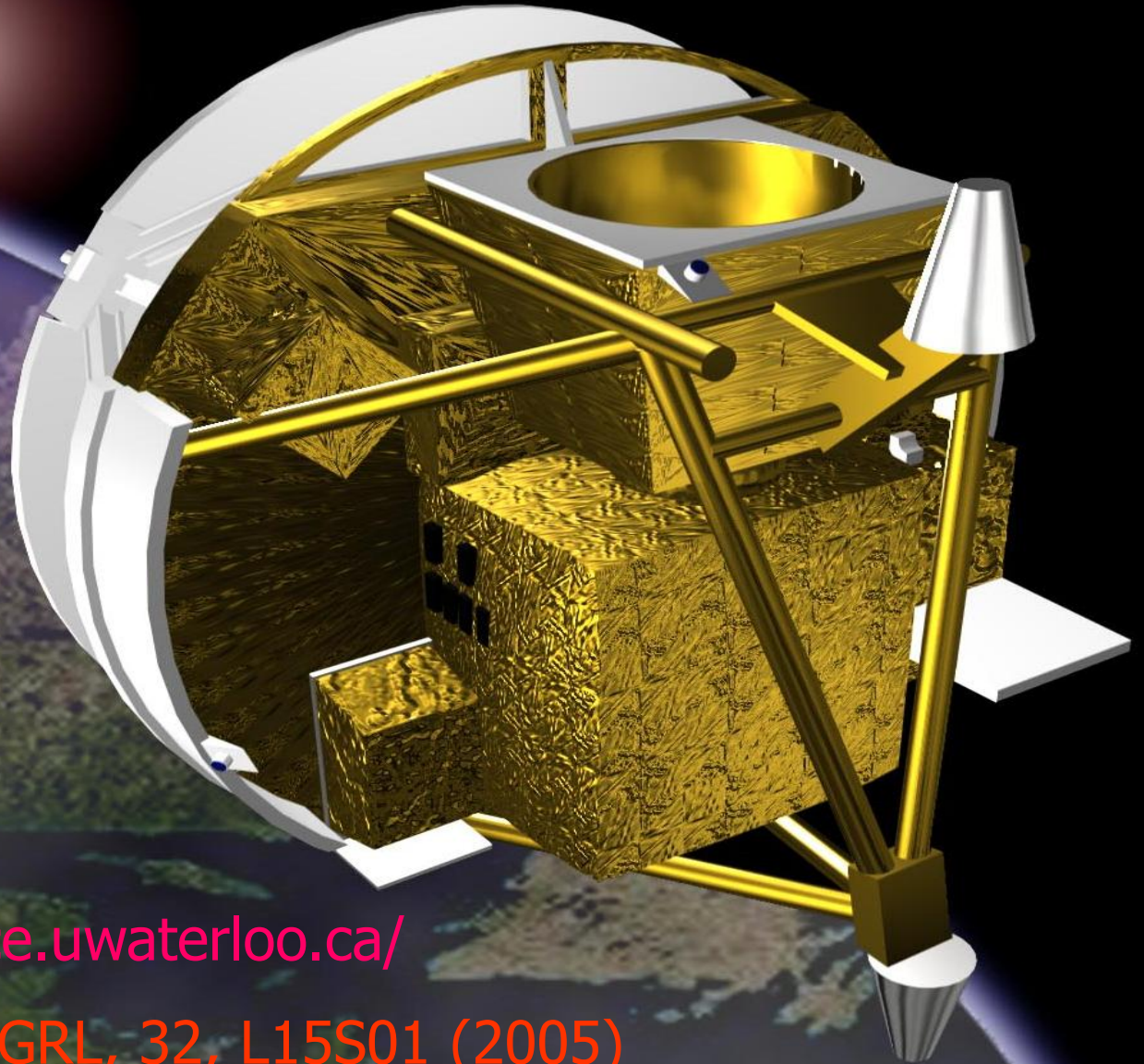
Then:

1. Co-I on proposed NASA ATMOS follow-on
2. PI for a proposal to CSA for small IR nadir sounder (like IASI)
3. ACE



Infrared transmission at tangent altitudes between 15 and 90 km and between frequencies of 700 and 1100  $\text{cm}^{-1}$  (recorded using the ATMOS interferometer onboard the space shuttle, Feb. 1992)

# ACE Satellite



<http://www.ace.uwaterloo.ca/>

Bernath et al., GRL, 32, L15S01 (2005)



# Instruments

- Infrared Fourier Transform Spectrometer (FTS) operating between 2 and 13 microns ( $750\text{-}4400\text{ cm}^{-1}$ ) with a resolution of  $0.02\text{ cm}^{-1}$
- 2-channel visible/near infrared Imagers, operating at 0.525 and 1.02 microns (cf., SAGE II)
- UV / Visible spectrometer (MAESTRO) 0.4 to 1.03 microns, resolution  $\sim 1\text{-}2\text{ nm}$  (added after initial selection)



# Central Canada- NE USA



CSA acted as "prime contractor"

FTS, ABB-  
Bomem Inc

DFL, T-Vac  
testing; S/C  
assembly

CSA HQ, MOC

FTS testing

ACE SOC  
Waterloo, ON, Canada



# Timeline (“faster, cheaper”)

- **July 1997** AO by CSA for SCISAT-1 (Astro or Atm Sci)
- Jan. 1998 Proposal to Canadian Space Agency (CSA)
- Apr. 1998 Start of Phase A (feasibility) for 3 proposals
- Oct. 1998 Presentations and selection of ACE
- Feb. 1999 Project kick-off (Phase B: preliminary design)
- Oct. 1999 FTS Preliminary Design Review (PDR) (Phase C start: detailed design)
- Feb. 2001 FTS/Imager Critical Design Review (CDR) (Phase D start: build)
- Sept. 2002 Spacecraft integration & test
- Mar. 2003 FTS Instrument test (Toronto)
- May 2003 Final integration (DFL, Ottawa) (AIT)(SRR)
- **Aug. 2003** **Launch and LEOP (launch & early orbit phase)**
- Sept. 2003 Commissioning starts (Phase E: operations)
- Feb. 2004 Routine operations

Mission had a nominal 2-year lifetime – sixth anniversary Aug. 2009.

Key step: During Phase A engage the community in project.  
(Notice: no Pre-Phase-A studies!)

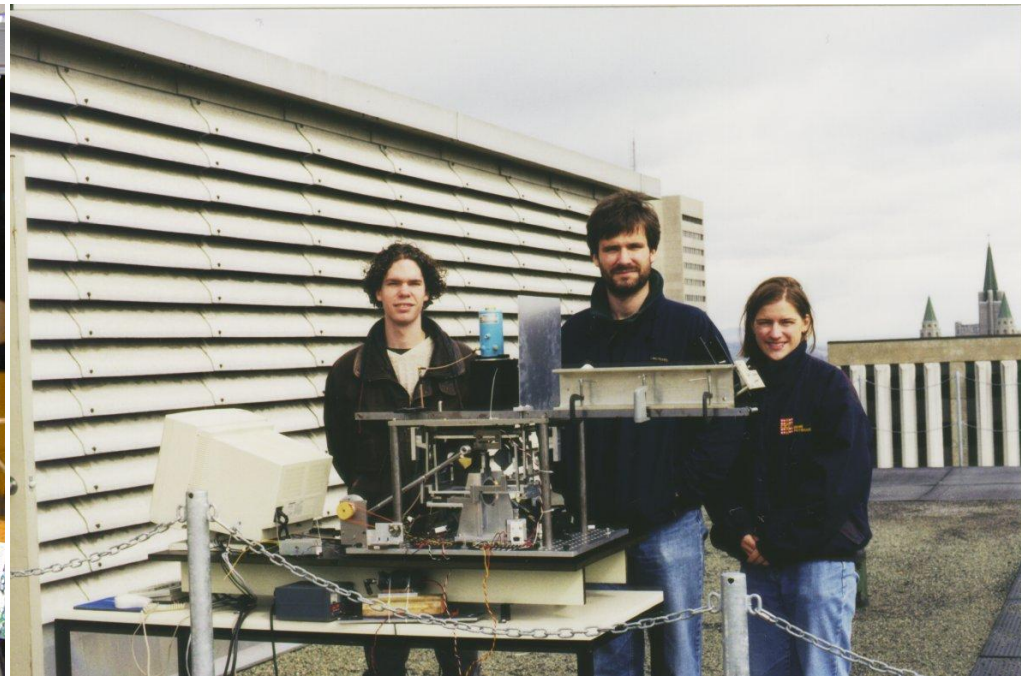
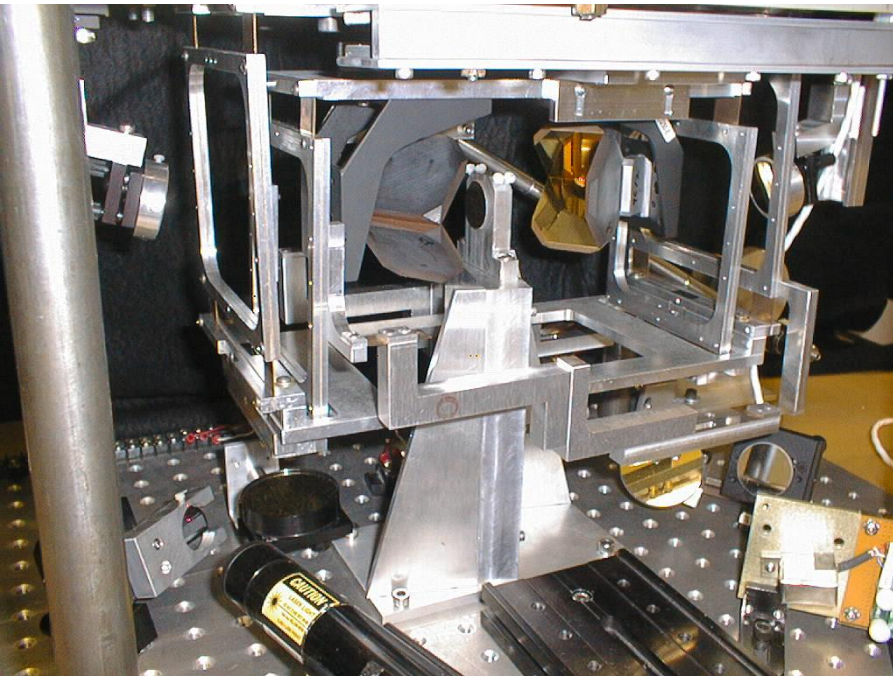




# Phase A FTS Breadboard

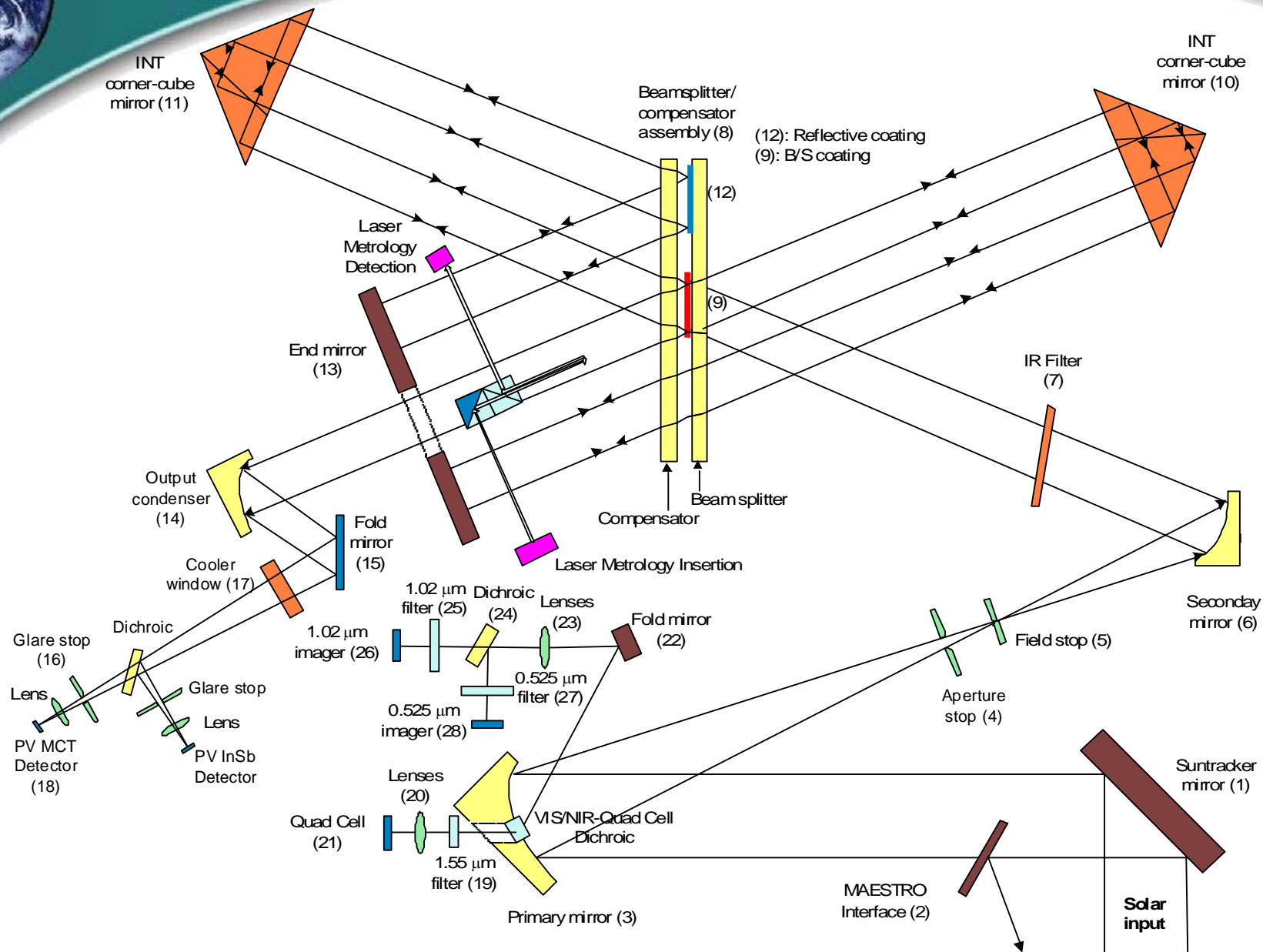
April 1999: Phase A (feasibility study) starts, but we were scientists and FTS users, not hardware designers.

1. Design work started at ABB-Bomem (our FTS industrial partner in Quebec City)
2. Possible prototype built by students of Pierre Tremblay (U. Laval); although their design was not used for flight, the breadboard was present at selection meeting in Oct. 1999



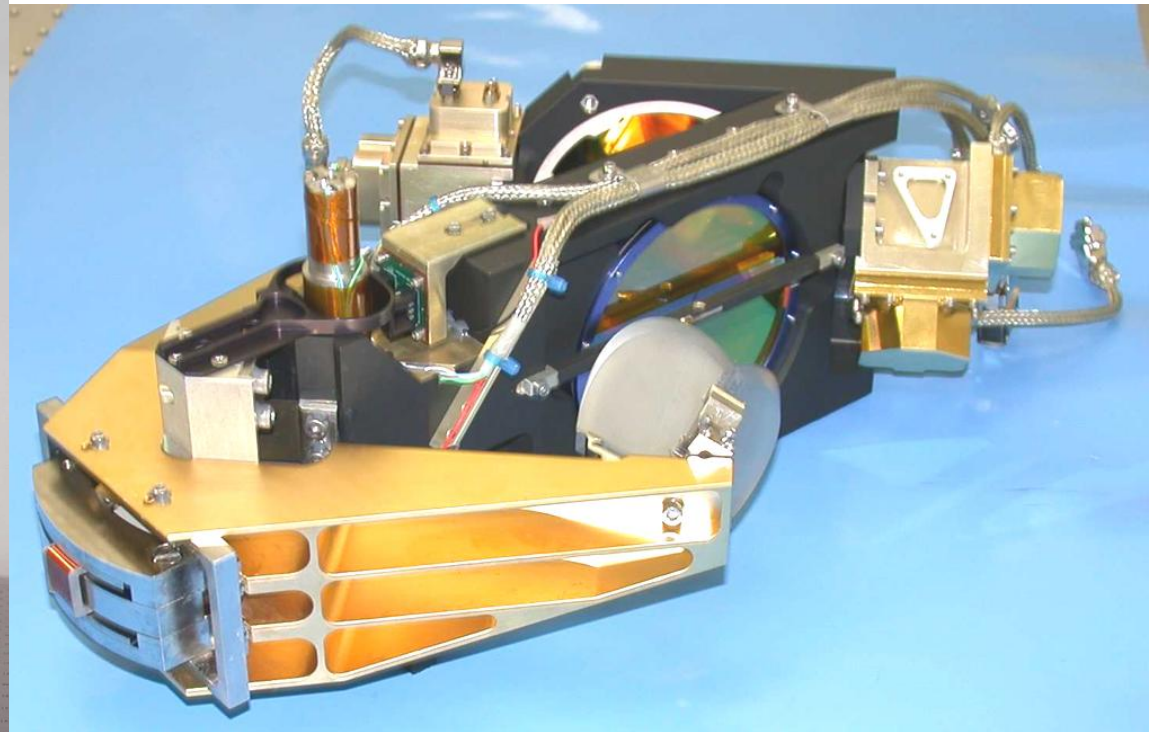


# Optical Layout (ABB-Bomem, Quebec City)





# ACE-FTS (Flight Model)



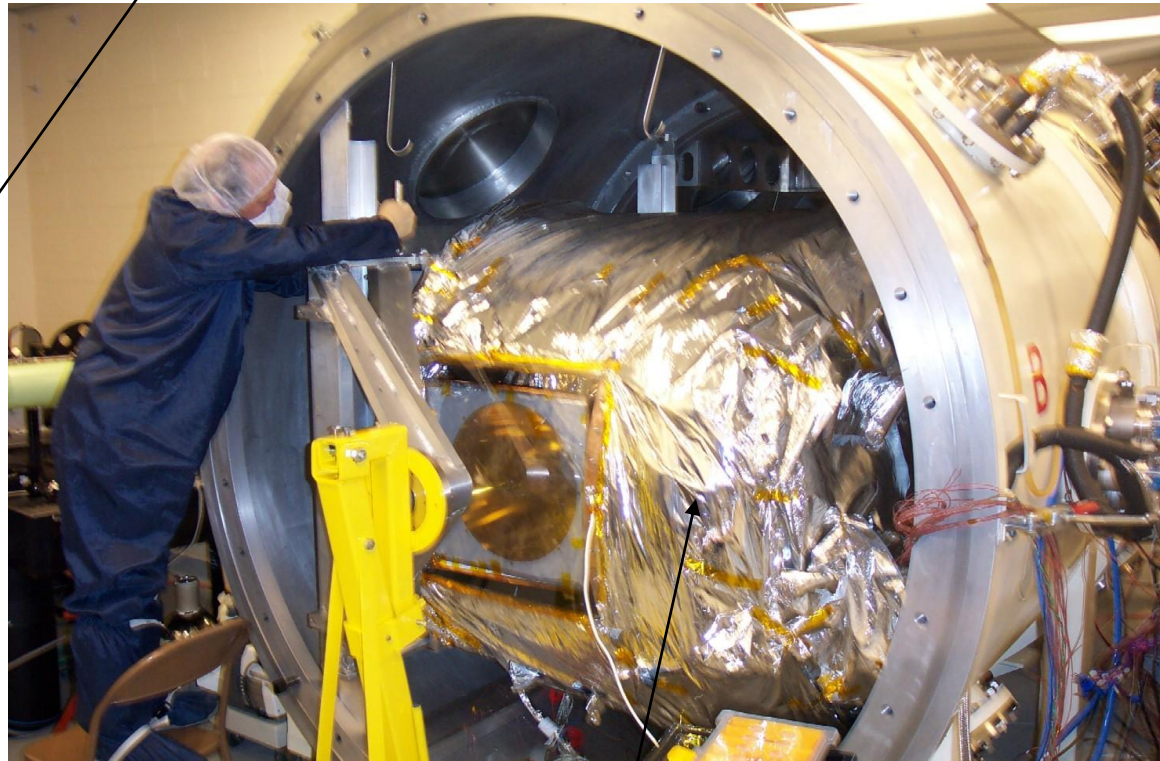
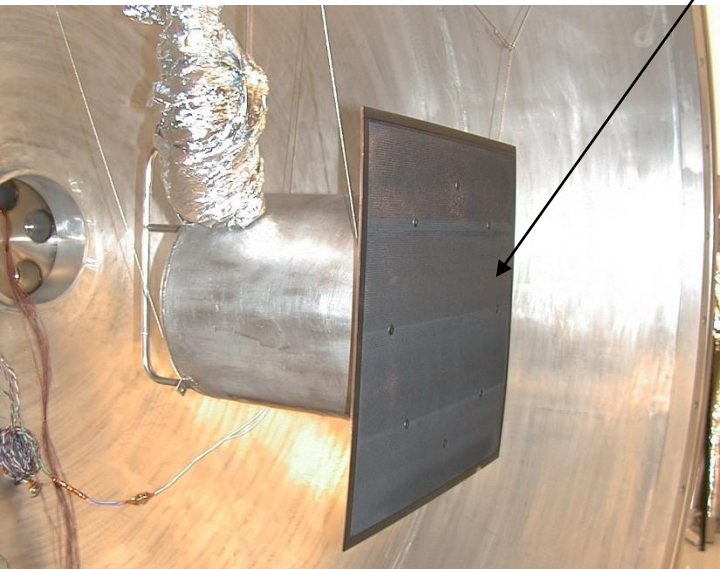




# Testing the FTS (U. of Toronto)

Testing of flight hardware by graduate students and post-docs

Detectors cooled to  $\sim 89$  K by directing the FTS passive cooler at the liquid He-cooled target

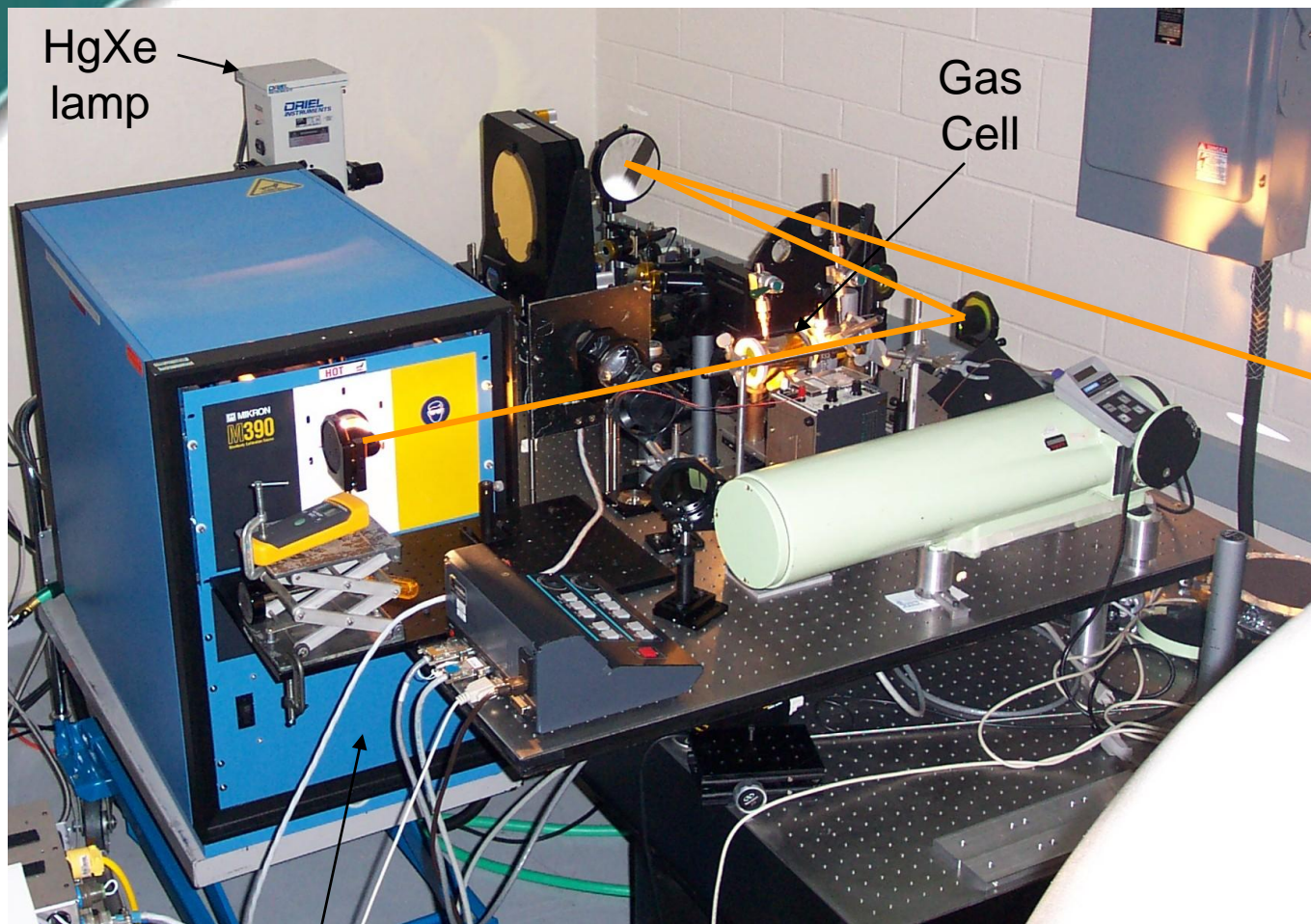


Multi-layer insulation (MLI)





# Gas Cell Measurements



3000°C Blackbody



to TVAC Window

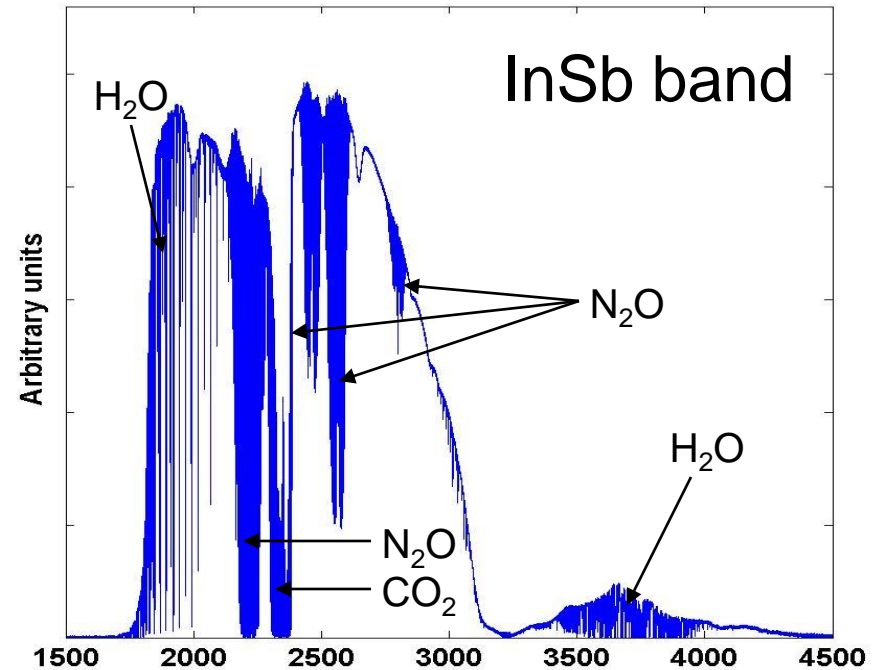
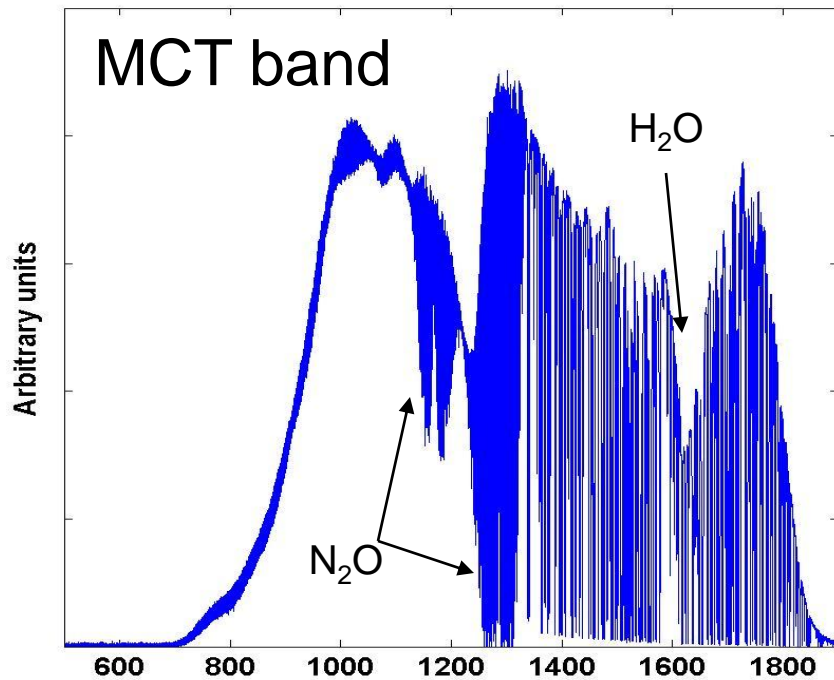


Gas Cell





# ACE-FTS Test Data & Results



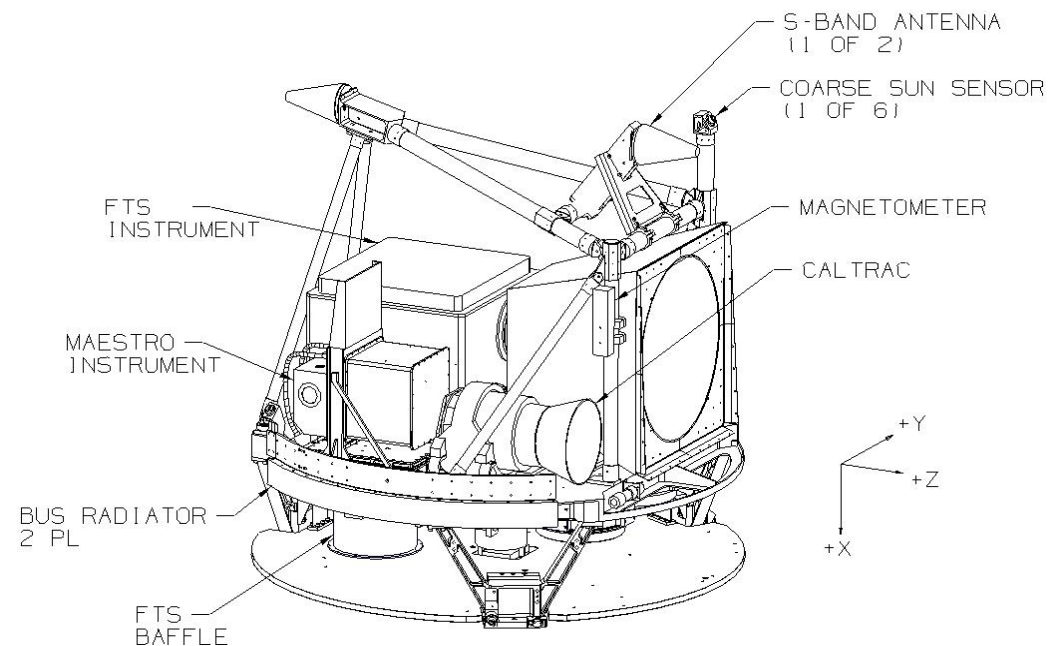
Detector 89 K, Instrument Nominal Temperature, ~12.0 Torr  $\text{N}_2\text{O}$

Instrument test motto: "Test as you fly"

# SCISAT-1 Spacecraft (Bristol Aerospace, Winnipeg)

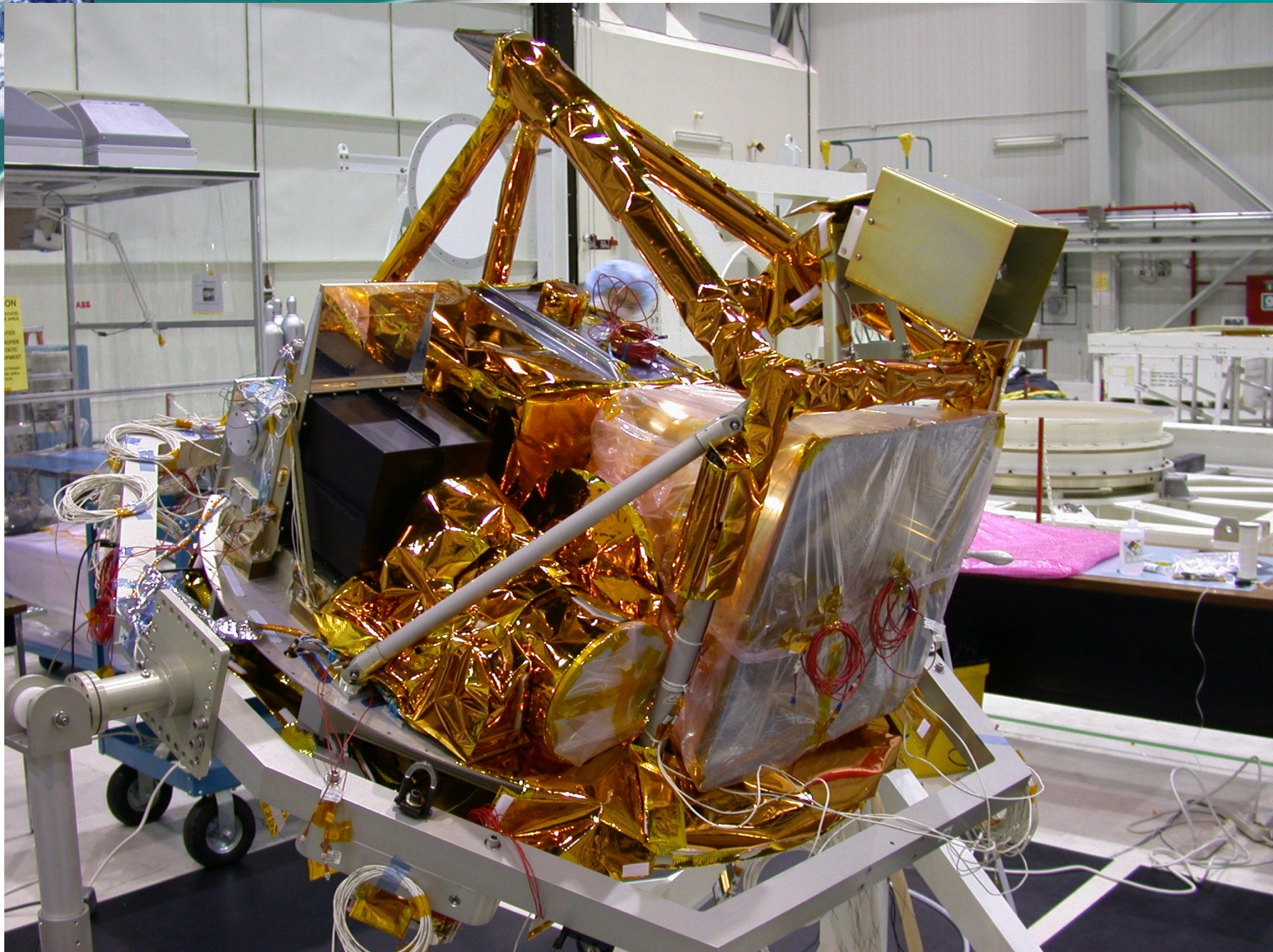
## Key Specifications

Spacecraft diameter	112 cm
Spacecraft mass	152 kg
Pointing control:	
Pitch/yaw ( $3\sigma$ )	$\pm 0.2^\circ$
Roll ( $3\sigma$ )	$\pm 0.7^\circ$
Pointing knowledge:	
Pitch/yaw	$\pm 0.1^\circ$
Roll	$\pm 0.6^\circ$
Solar array EOL power	175 W
Spacecraft OA power	75 W
Bus reliability (2-yrs)	80 %
Mass memory	1.5 Gb





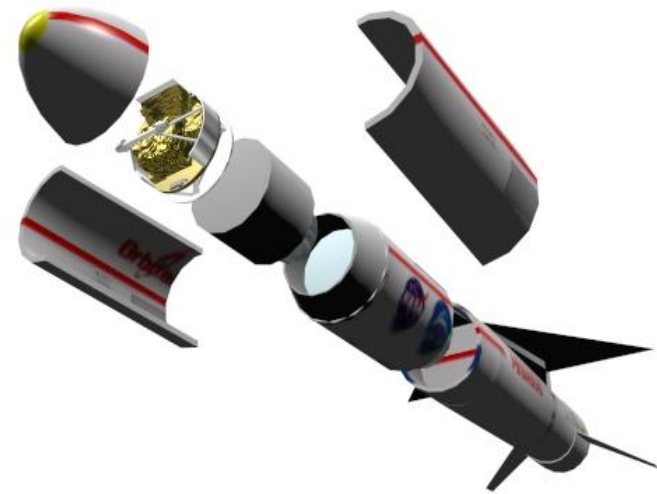
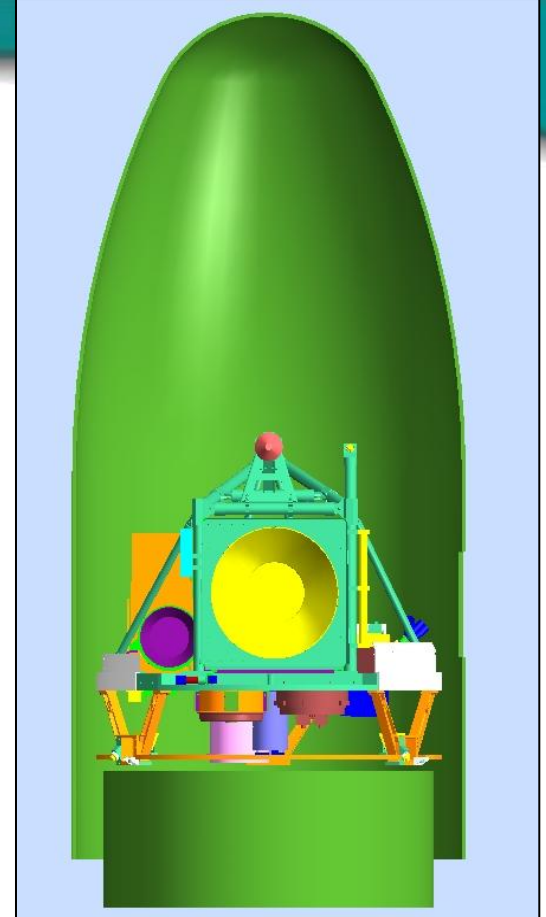
# FTS Integration to S/C Bus (DFL Ottawa)





# SCISAT-1 Launch

- Pegasus XL launch vehicle (provided by NASA)
- Launched from Vandenberg Air Force Base
- Nominal orbit of 650 km and 73.9° inclination
- Unpowered launch with S/C to roll slowly on orbit





# Integration of S/C to Pegasus Rocket





# Pegasus XL Launch Vehicle



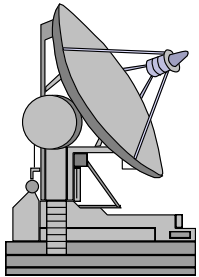




# Ground Segment



SCISAT-1



St. Hubert

- Primary Ground Station



University of Waterloo

- Science Operations Centre



Saskatoon

- Secondary Ground Station

Fairbanks, USA

Alaska Satellite Facility (NASA)

Kiruna, Sweden

ESA Satellite Station

## Communications

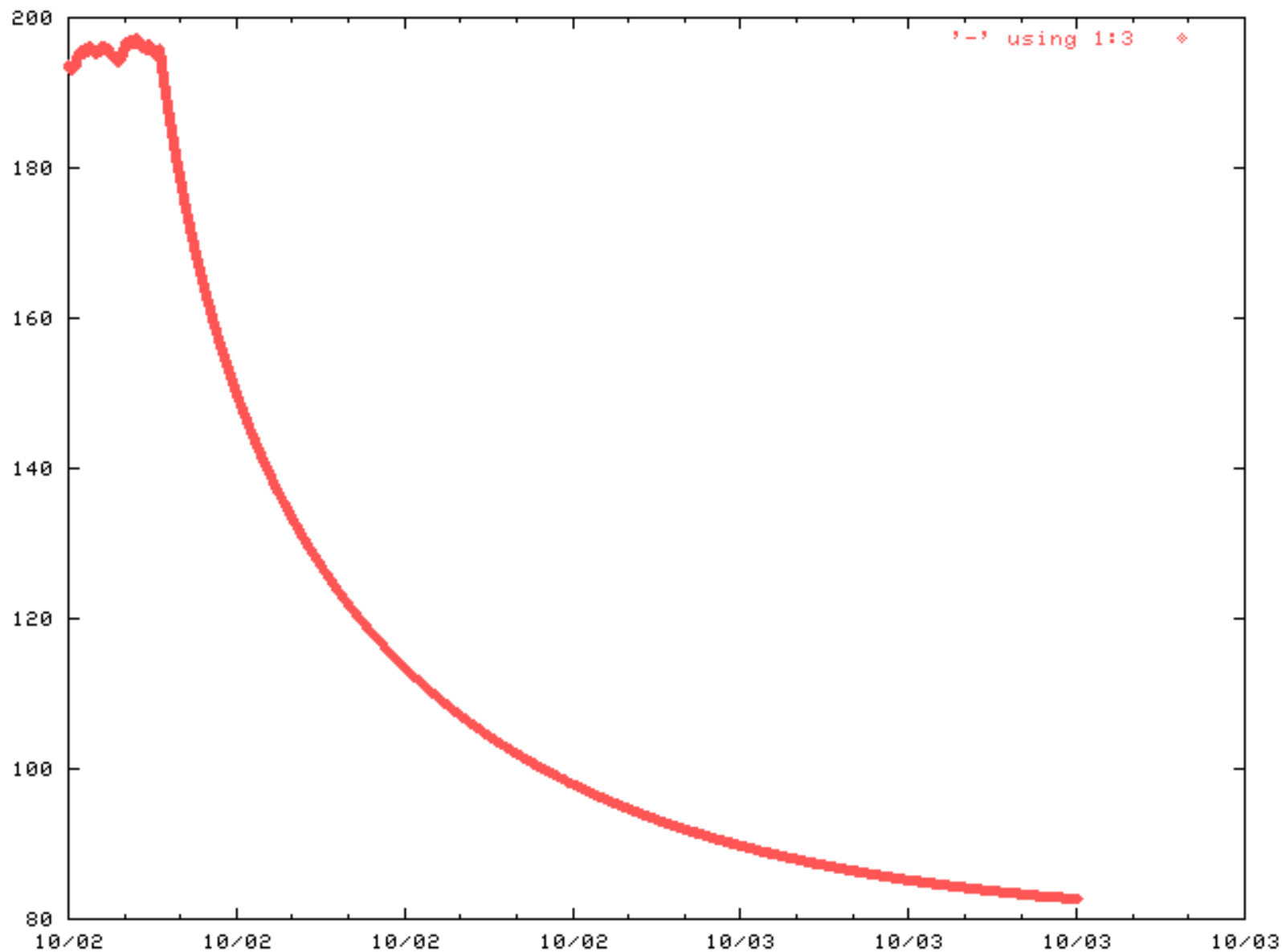
- Downlink
  - S-band
  - CCSDS multiplexing
  - 40 kbps to 4 Mbps
  - $10^{-9}$  BER
- Uplink
  - S-band
  - CCSDS command protocol
  - 4 kbps

## Tracking

- 2-way Doppler range-rate measurements



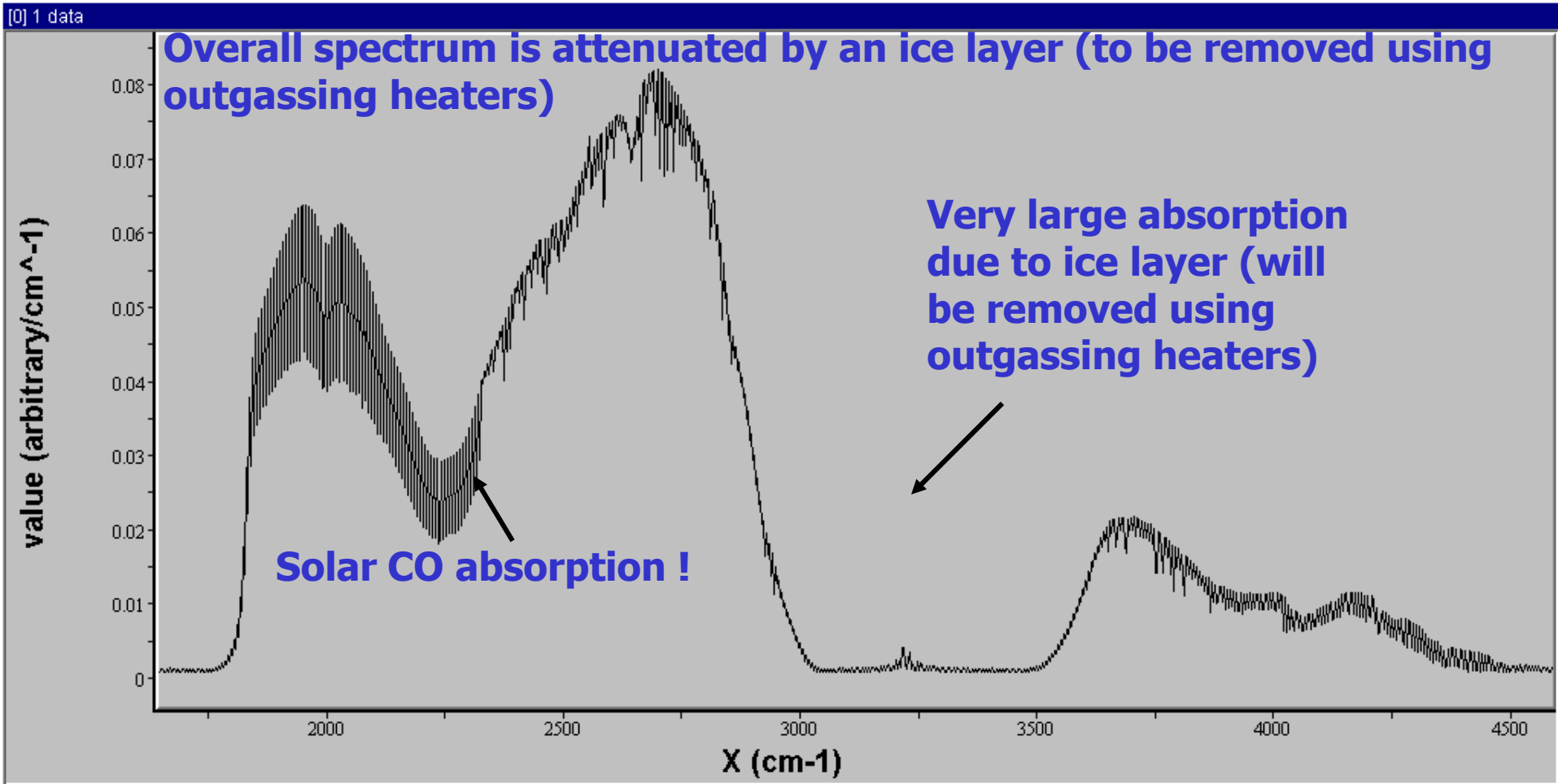
# FTS Detector Cool-Down-Oct. 2003 (roll control)







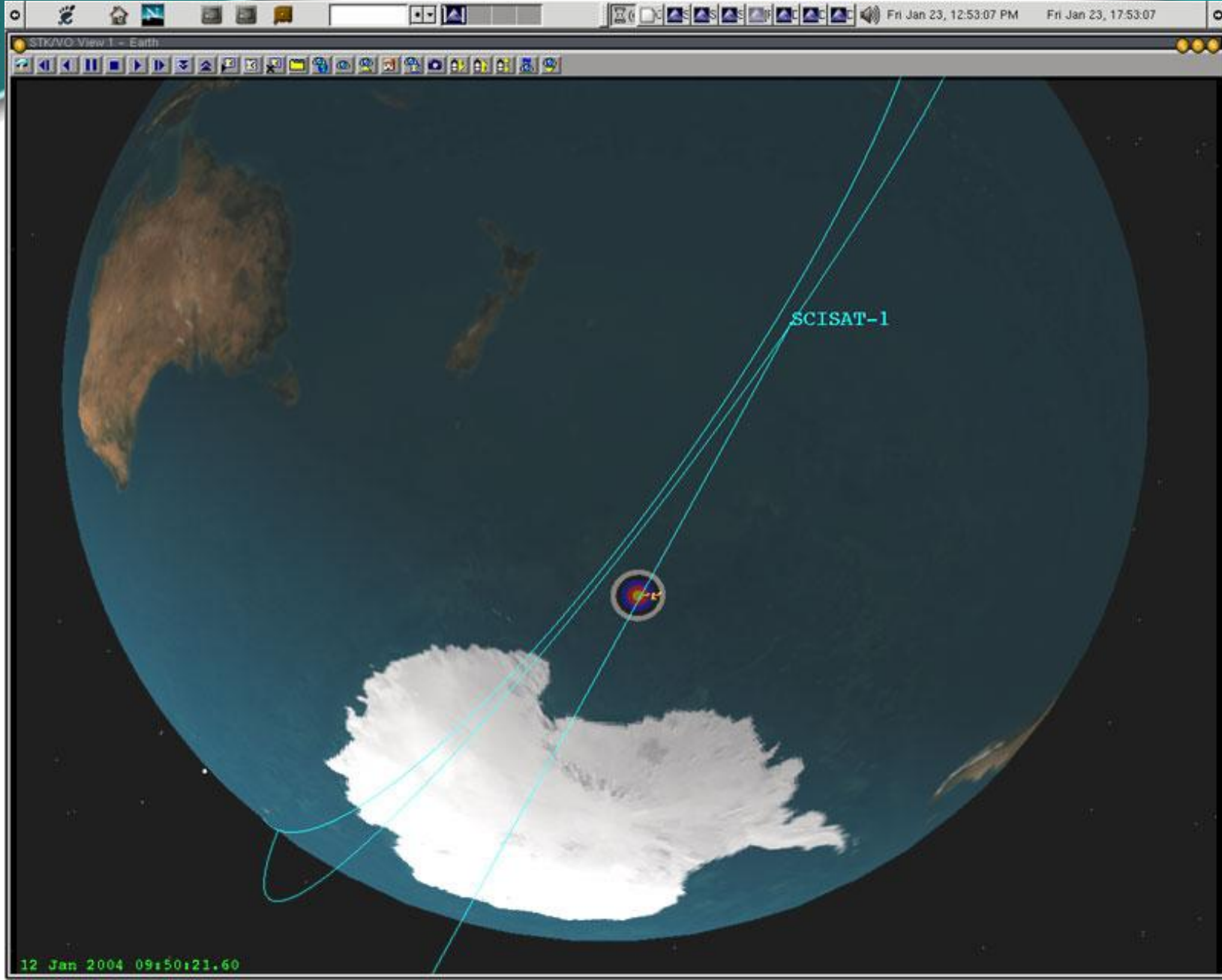
# "First Light": InSb Detector Spectrum-Nov. 2003



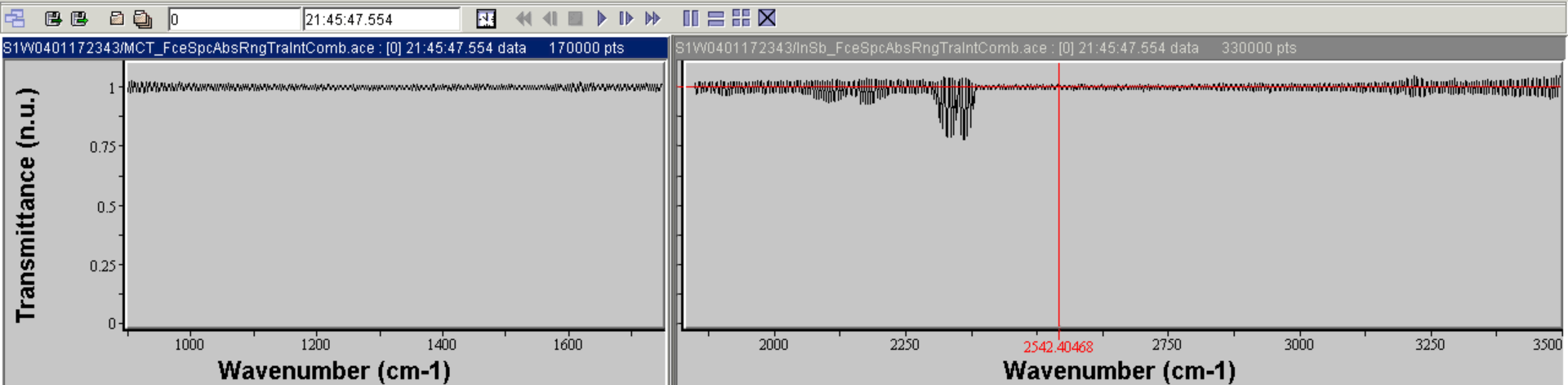
**Note: exo-atmospheric data (pure Sun) with absorption due to many atomic lines and CO (IR Fraunhofer spectrum)**



# Jan. 12, 2004 9:50:23 (UTC)



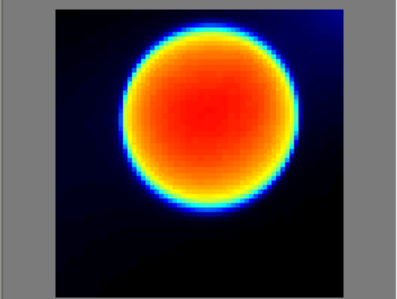




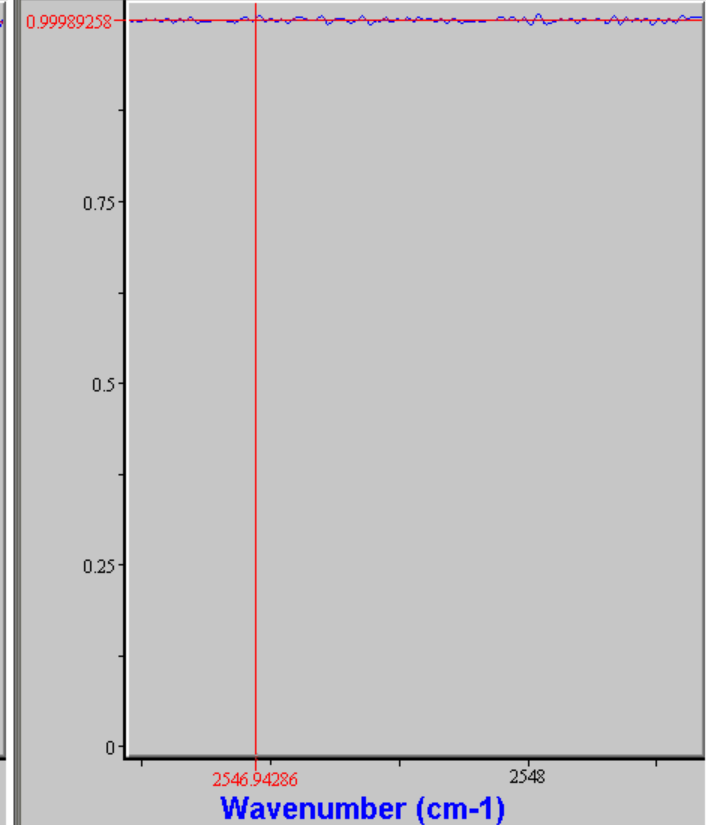
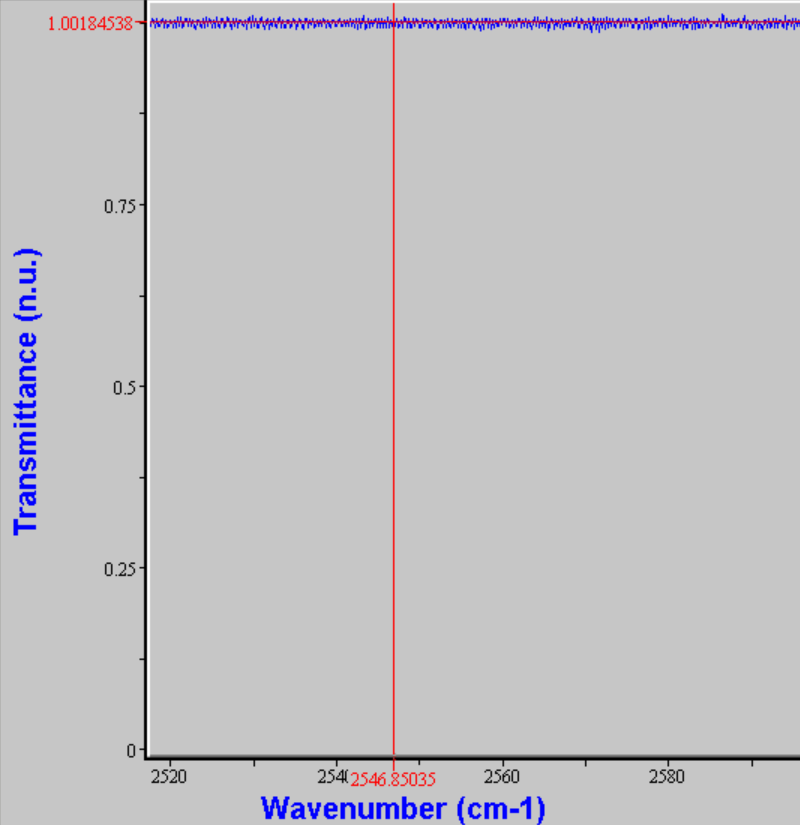
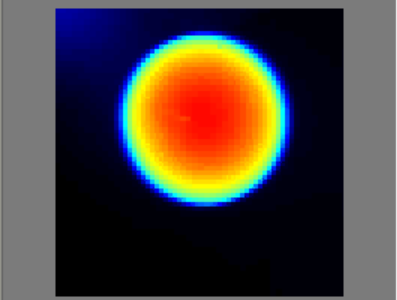
S1W0401172343/nir.ace [151] 21:45:47.750 d...

S1W0401172343/InSb\_FceSpcAbsRngTraIntComb.ace : [0] 21:45:47.554 data 330000 pts

S1W0401172343/InSb\_FceSpcAbsRngTraIntComb.ace : [0] 21:45:47.554 data ...



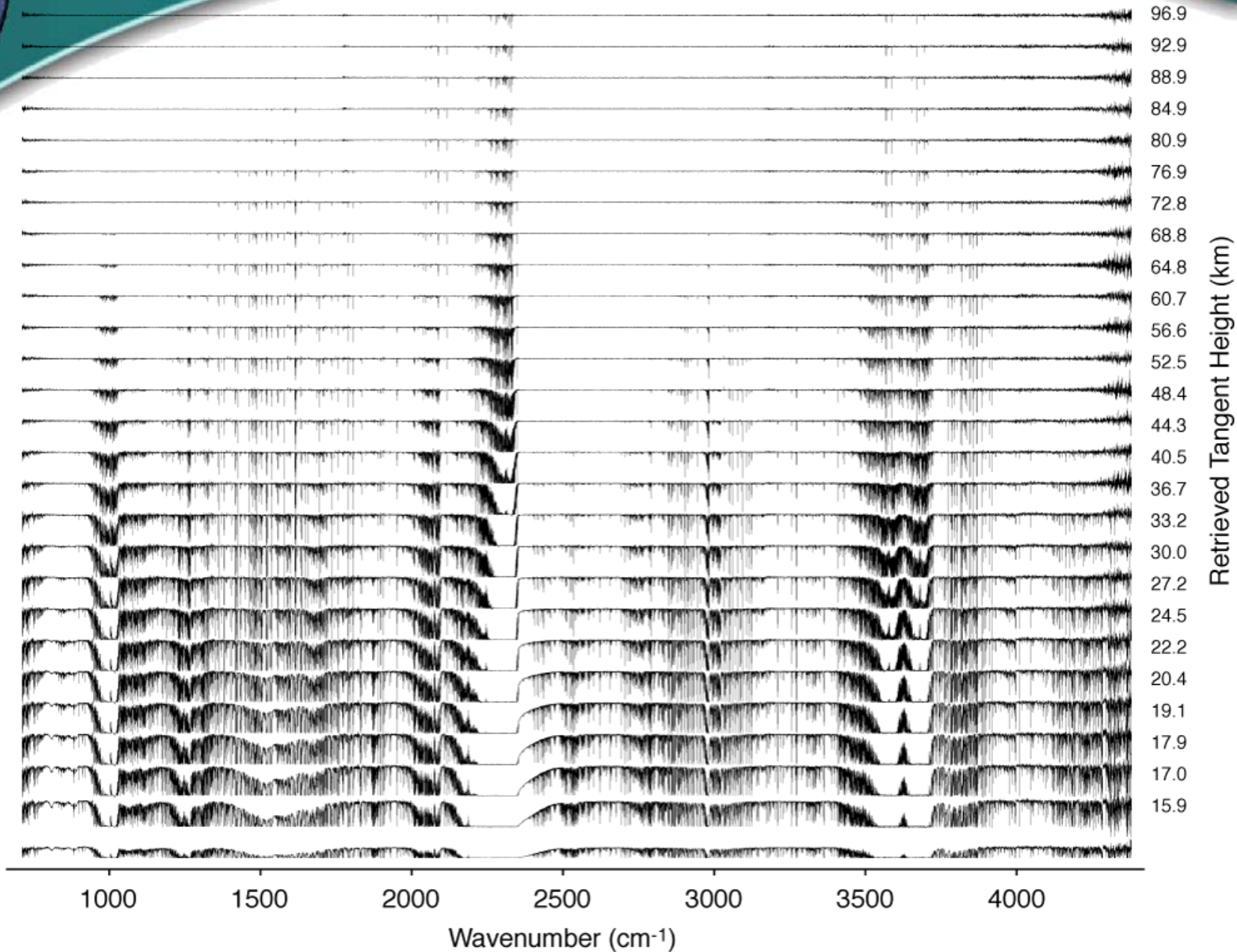
S1W0401172343/vis.ace [151] 21:45:47.750 d...





# Occultation sequence

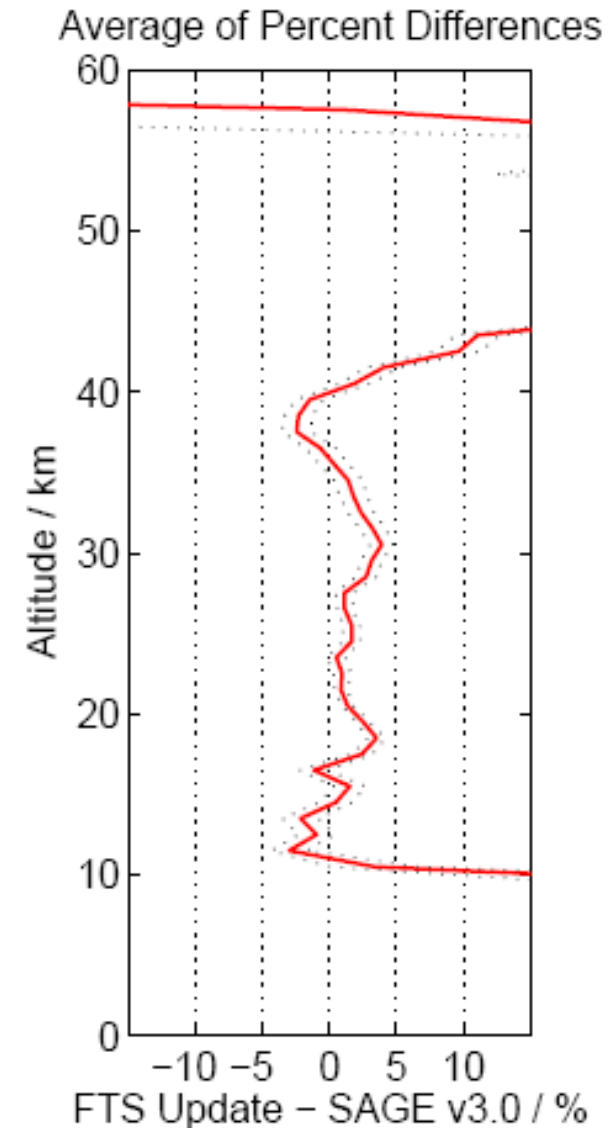
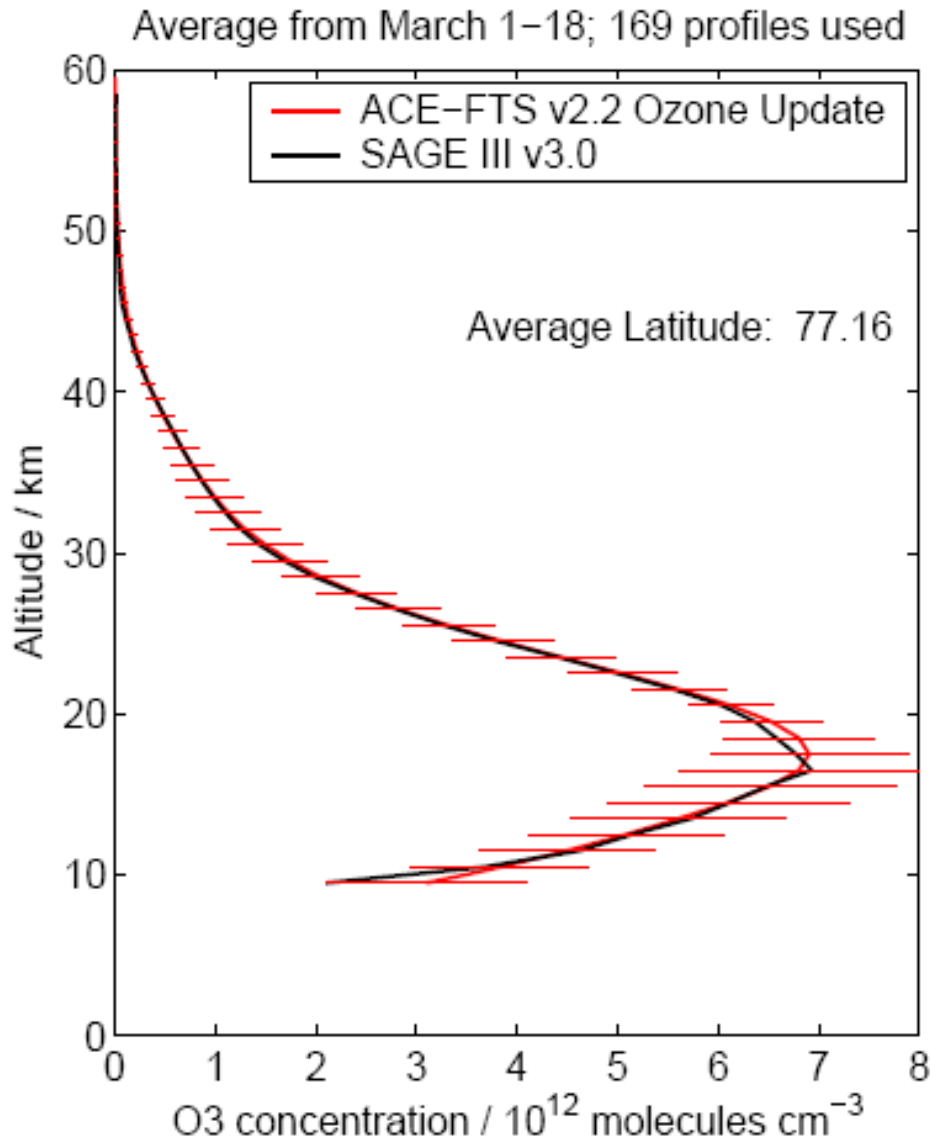
Atmospheric Transmittance







# Ozone ACE Cal/Val



K. Walker



# Science Goals (Verbatim from Phase A Report)

## **PRIORITY 1**

- Measurement of regional polar  $O_3$  budget to determine the extent of  $O_3$  loss. This will require measurements of  $O_3$ , tracers ( $CH_4$  and  $N_2O$ ), and meteorological variables (pressure and temperature).
- Measurement / inference of details of  $O_3$  budget by detailed species measurements (for  $O_3$ ,  $H_2O$ ,  $NO$ ,  $NO_2$ ,  $N_2O_5$ ,  $HNO_3$ ,  $HCl$ ,  $ClNO_3$ ,  $ClO$ ) and modelling.
- Measurement of composition, size and density of aerosols and PSCs in the visible, near IR and mid IR.
- Comparison of measurements in the Arctic and Antarctic with models to provide insight into the differences, with emphasis on the chlorine budget and denitrification.

## **PRIORITY 2**

- Mid-latitude  $O_3$  budget.
- Measurement of Arctic vortex descent.

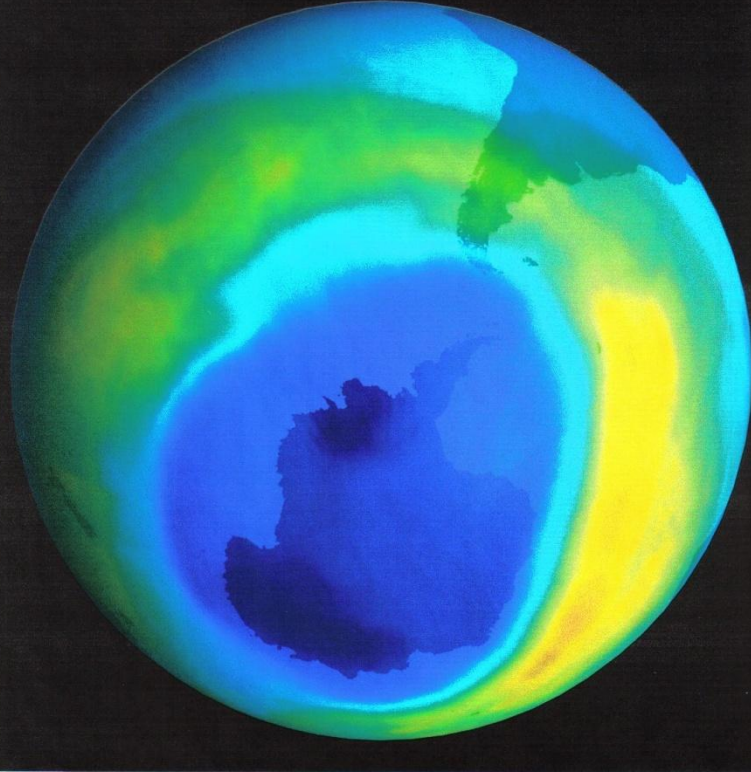
## **PRIORITY 3**

- Measurement of winds.
- Study of upper tropospheric chemistry.
- Monitoring of CFCs, CFC substitutes and greenhouse gases.



# Space Agency View of Mission Design

Ozone • September 6, 2000 • Total Ozone Mapping Spectrometer (TOMS)



Science problem  
(e.g., understand and  
measure ozone hole)

## Science Requirements Document

(e.g., ozone VMR with 5%  
accuracy from 10-50 km altitude)

**Instrument Requirements  
Document** (e.g., FTS with  
transmittance accuracy of 1%)

Subsystem hardware  
requirements



# Average CO<sub>2</sub> Profile 60-100 km

## CO<sub>2</sub> Mystery:

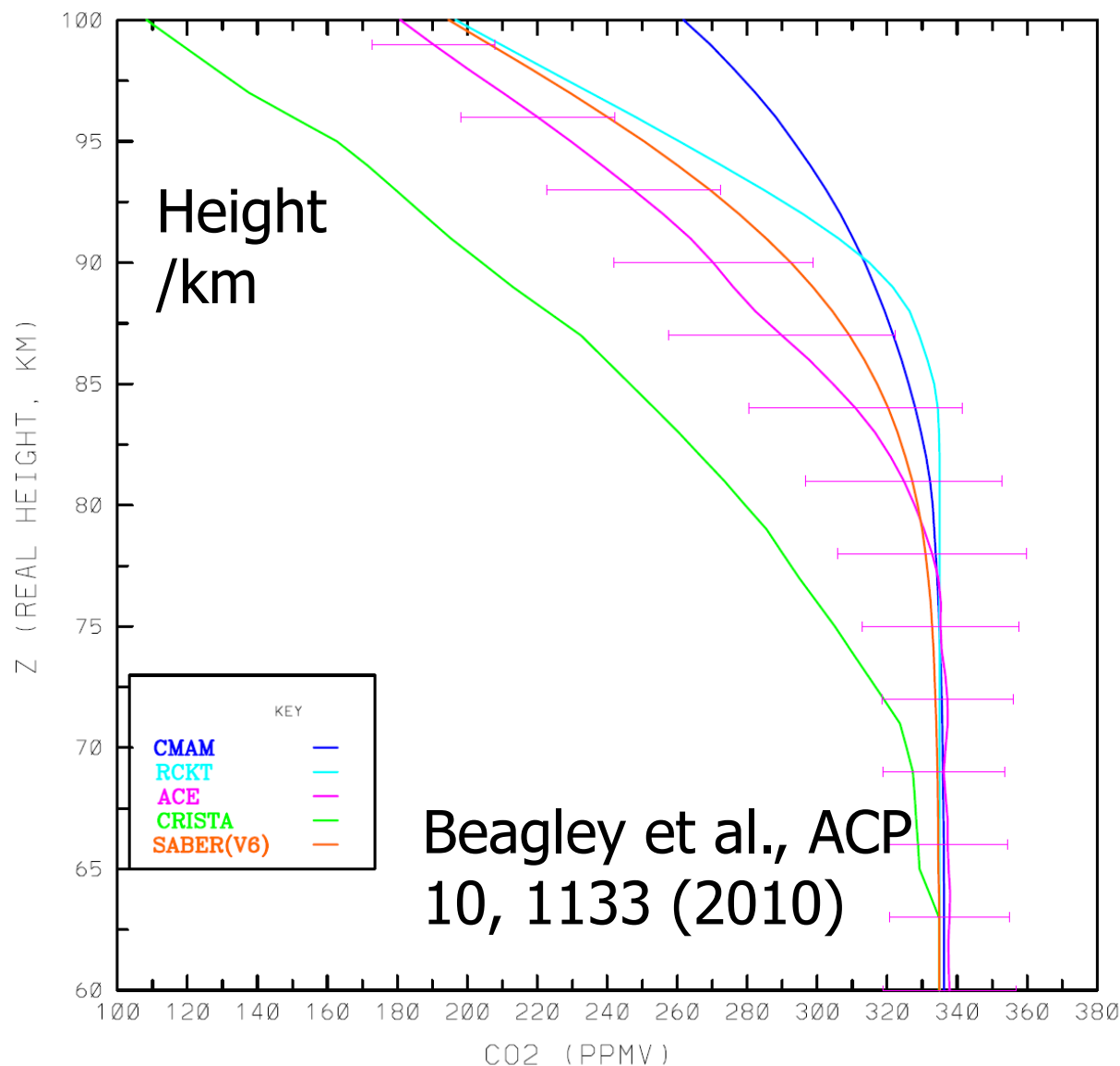
$\text{CO}_2 + h\nu \rightarrow \text{CO} + \text{O}$

ACE profile (purple)  
does not agree with  
CMAM model (blue)

Good agreement for  
CO, however.

Where has CO<sub>2</sub> gone?

Perhaps there are  
reactions with  
meteoritic dust?





# Atmospheric Dynamics: Asian Summer Monsoon Anticyclone

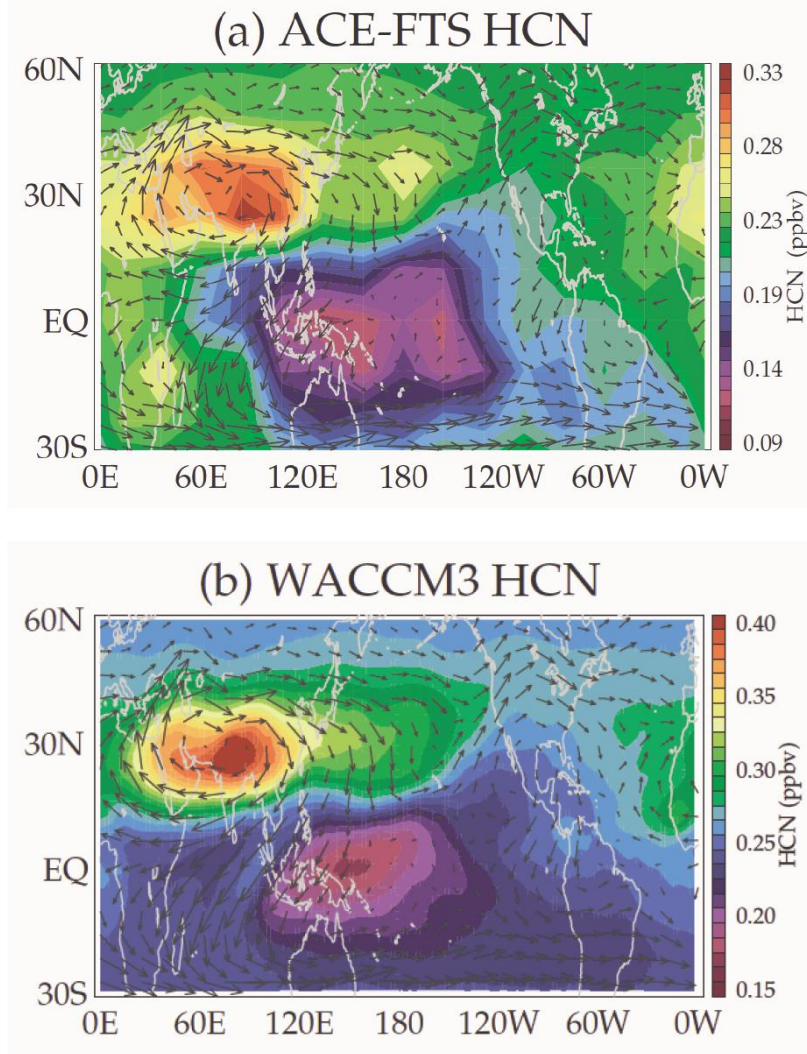


Figure 1. Time average mixing ratio (ppbv) of hydrogen cyanide (HCN) near 13.5 km during boreal summer (June-August) derived from (a) ACE-FTS observations during 2004-2008, and (b) WACCM chemical transport model calculations. Arrows in both panels denote winds at this level derived from meteorological analysis, showing that the HCN maximum is linked with the upper tropospheric Asian monsoon anticyclone.

At 13.5 km in altitude, HCN is high over Tibet because of rapid lofting of polluted air from India and China during summer; it is lower over the Pacific Ocean because of deposition and destruction in the ocean. Randel et al., *Asian monsoon transport of pollution to the stratosphere*, Science (in press)



# Some Lessons Learned

1. Don't micromanage- a professional manager is needed (lots of paper work!)
2. Sign up partners from all sectors as early as possible – industry, government labs (e.g. weather service, etc.), other universities, especially international partners with money (e.g., Belgians and NASA)
3. Be open and inclusive (and avoid internal competition) - personal factors are important
4. Be very suspicious of claims of suppliers
5. Engineers work to requirements (not the same as scientists!)
6. Good acronym, well written proposal, timely science, favourable politics, etc.





# ACE Partners (Selected)

- Canada- K. Walker, J. Drummond, K. Strong, J. McConnell, W. Evans, T. McElroy, I. Folkins, R. Martin, J. Sloan, T. Shepherd, T. Llewellyn, etc.
- USA- NASA launched ACE: C. Rinsland, L. Thomason (NASA-Langley), C. Randall (U. Colorado), M. Santee, L. Froidevaux, G. Toon, G. Manney (JPL), etc.
- Belgium- supplied CMOS imager chips: R. Colin, P.-F. Coheur, M. Carleer (ULB), D. Fussen, M. DeMaziere (IASB), M. Mahieu, R. Zander (Liege), etc.
- UK- J. Remedios (Leicester), P. Palmer (Edinburgh), M. Chipperfield (Leeds), etc.
- France- C. Camy-Peyret, C. Clerbaux, C. Brogniez, G. Dufour, D. Hauglustaine (Paris)
- Japan- M. Suzuki (JAXA), Y. Kasai
- Sweden- G. Witt (Stockholm)

Funding: CSA, NSERC, NERC



**National Centre for  
Earth Observation**

NATURAL ENVIRONMENT RESEARCH COUNCIL