

# Emerging Technologies Challenge Workshop

## EO Technology Needs at ESA

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04/05/2017

# Content



## Overview of Earth observation Programmes:

- Meteorological & EUMETSAT
- Copernicus & EU
- Earth Explorers

## Future

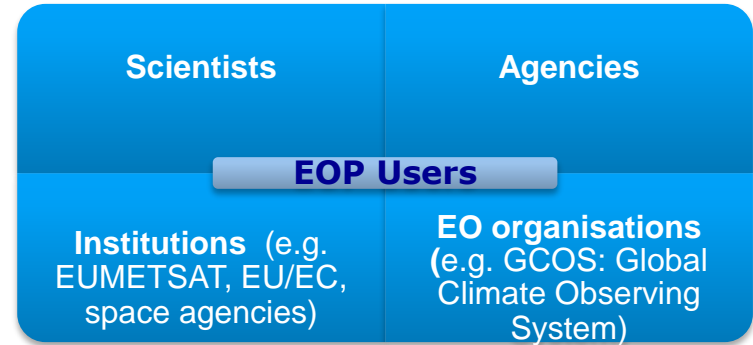
- ESA Technology developments for EO
- mission concepts : Compendium overview



# ESA Earth Observation Programmes



ESA Earth Observation (EO) activities are **user-driven** and implemented through **optional EO Programmes (EOP)**



EOP include **end-to-end** activities, i.e. from conception to exploitation of space missions, for different types of missions:

- **RESEARCH MISSIONS** → **Earth Explorers** (Core, Opportunity, Fast-Track), for R&D and demonstration of EO techniques prior to their “operationalisation”
- **EARTH APPLICATIONS MISSIONS** → **Earth Watch**, i.e. (pre-)operational missions in partnership, which can be
  - **Earth Watch SERVICE type**: partners EUMETSAT, EU / EC,..
  - **Earth Watch PRIVATE INITIATIVE**: with specific Member States, industry,..

# ESA Earth Observation Missions



## Living Planet

### Earth Explorer

Research driven

### Earth Watch

Operational Service driven

#### Core Missions

**GOCE**  
2009-2013

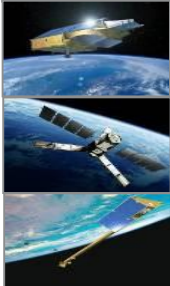
**ADM-Aeolus**  
Dec 2017

**EarthCARE**  
2019

**Biomass**  
(EE7) 2021

Call EE10 – (2017)

#### Opportunity



**CryoSat-2**  
8 April 10

**SMOS**  
2 Nov 09

**Swarm**  
22 Nov 13

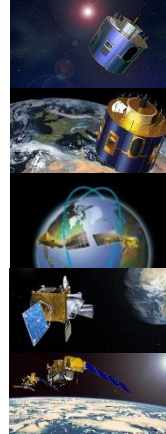
#### Fast Track



**FLEX (EE8)**  
2023

Call EE9 – (2017)

#### Meteorology



**Meteosat**

**MSG**

**MetOp**

**MTG**  
(2 s/c x 3)  
**MetOp SG**  
(2 s/c x 3)

#### Copernicus



**Sentinel 1 a/b/c/d**

**Sentinel 2 a/b/c/d**

**Sentinel 3 a/b/c/d**

**Sentinel 4**  
(on MTG)

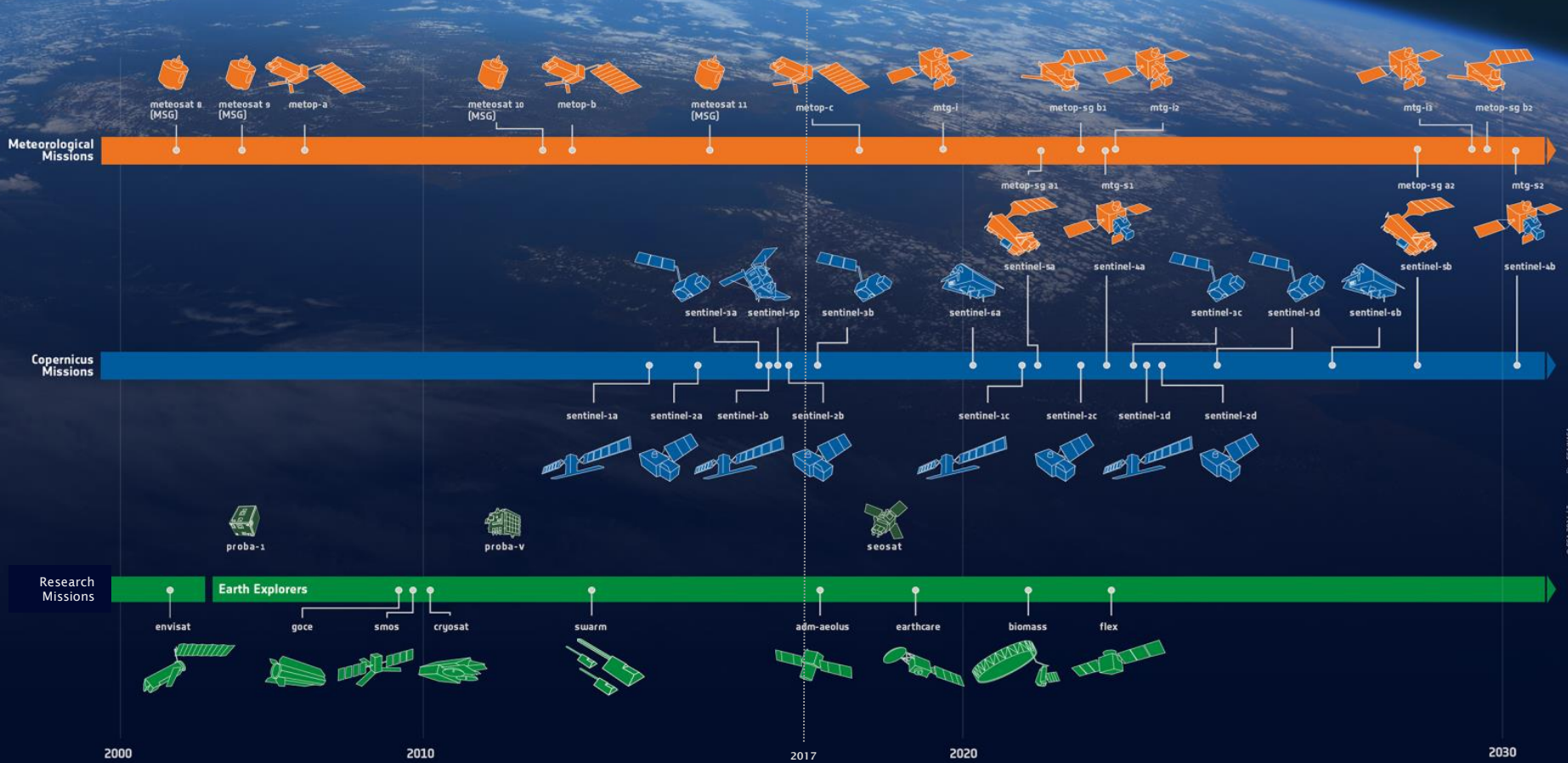
**Sentinel 5 precursor**

**Sentinel 5**  
(on MetOp SG)

**Sentinel 6 (Jason)**



# → ESA-DEVELOPED EARTH OBSERVATION MISSIONS



# Outlook of Earth Watch programmes with Eumetsat



Developed in cooperation with ESA's partner, Eumetsat, as Europe's contribution to the World Meteorological Organization's space-based Global Observing System:

**Meteosat Second Generation** (2002, 2005, 2012, 2015) – series of four satellites providing images of Earth from geostationary orbit.

**Meteosat Third Generation** (2021– ) series of six geostationary satellites providing images (four satellites) and atmospheric sounding (two satellites).

**MetOp** (2006, 2012, 2018) – series of three satellites providing operational meteorological observations from polar orbit.

**MetOp Second Generation** (2021– ) two series of polar-orbiters, three satellites in each series, continuing and enhancing meteorological, oceanographic and climate monitoring observations from the first MetOp series.

# Outlook of Earth Watch programmes with Eumetsat...the future



EUMETSAT will prepare the user consultation process that needs to be initiated with ESA in the 2025-2027 timeframe as the first formal step in the planning of future programs through an appropriate balance between observations from geostationary and low Earth orbits.

**Meteosat Fourth Generation** (2030-) geostationary observation.

**MetOp Third Generation** (2040- ) polar-orbiters.

**More integration likely between EUMETSAT and EU assets** (as for S-4/5, S6)

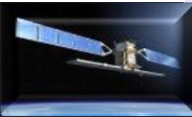


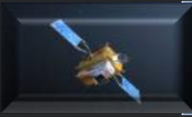
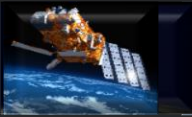

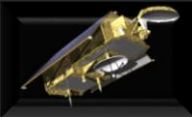
# Outlook of Earth Watch programmes with EU/EC : Copernicus Missions (1/2)

- Copernicus is a European space flagship programme
- It is led by the European Union. ESA coordinates the space component
- Copernicus provides the necessary data for operational monitoring of the environment and for civil security
- Copernicus missions include Sentinels and contributing missions (from national Agencies and companies)



# Copernicus : actual sentinels



	Sentinel	Launch	Instrument	Application	Downlink	Note
	<b>Sentinel-1 A</b> <b>Sentinel-1 B</b>	3 April 14 25 April 16	C-Band SAR	All weather, day/night applications, interferometry	X-Band Opt. → EDRS	polar-orbiting satellites
	<b>Sentinel-2 A</b> <b>Sentinel-2 B</b>	22 June 15 7 March 17	Multispectral	Land applications: urban, forest, agriculture,	X-Band Opt. → EDRS	polar-orbiting satellites
	<b>Sentinel-3 A</b> <b>Sentinel-3 B</b>	16 Feb 16 Plan 2017	C/Ku Radar Altimeter 23.8/36.5 GHz radiometer OLCI spectrometer VIS-SWIR-TIR radiometer	Wide-swath ocean color, vegetation, sea/land surface temperature, altimetry	X-Band	polar-orbiting satellites
	<b>Sentinel-4 A</b> <b>Sentinel-4 B</b>	plan 2020 plan 2030	UVN Spectrometer Infrared Sounder	Atmospheric composition monitoring, trans-boundary pollution	26 GHz	Payload to be embarked on MTG (GEO).
	<b>Sentinel-5 A</b> <b>Sentinel-5 B</b>	plan 2022 plan 2028	UVNS Spectrometer Infrared Sounder Multi-view/pol Imager	Atmospheric composition monitoring,	26 GHz	Payload to be embarked on Metop-SG (LEO).
	<b>Sentinel-5 P</b>	June 2017	TROPOMI (passive imaging spectrometer)	Tropospheric composition monitoring	X-Band	to follow ground track of Suomi-NPP
	<b>Sentinel-6 A</b> <b>Sentinel-6 B</b>	plan 2021 plan 2025	ESA Instruments: C/Ku Radar Altimeter and GNSS Radio-Occultation	Sea-level, wave height and marine wind speed	X-Band	partnership between EU/ESA/EUMETSAT and the NOAA/NASA

# Copernicus Evolution : Mission Concepts & Thematic Areas

(proposed as High Priority from the EC, **Current status, all TBD/TBC**):

- CO<sub>2</sub> Monitoring
  - Polar ice/ocean interfer. altimetry
  - Other polar (Arctic) observations
  - Land thermal imaging
  - Soil moisture, ocean salinity
  - Hyperspectral land imaging
- climate change**
- marine & polar envir. monitoring**
- land monitoring (agriculture, food security)**  
**- emergency management (geohazards)**  
**- marine envir. monitoring**



# Copernicus evolution: Candidates

(concepts + technology - **Current status , all TBD/TBC**)



- Several key observation gaps already known. **Examples:**
  - monitoring of greenhouse gases like CO<sub>2</sub> (emphasised by EC)
  - high-resolution thermal infrared imaging (on Sent-2 no TIR imager like Landsat)
- Other gaps quite evident from evolution of user needs. **Examples:**
  - new data for polar regions,
  - persistent (from GEO) rapid response imaging, ..
- Role expected for follow-on's of already demonstrated EO techniques. **Examples:**
  - interferometric SAR altimetry (CryoSat),
  - hyperspectral imaging (under National Developments),
  - SMOS follow-on
  - Next Generation Gravity Mission

# Copernicus Program Evolution

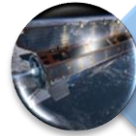


EO Capability	Actual Sentinels	Extended Data Availability	Sentinels Expansion	Second Generation	Instrument	Orbit
Earth Microwave (3D) Imaging	Sentinel-1 A Sentinel-1 B	Sentinel-1 C Sentinel-1 D	Sentinel-1 SG	Sentinel-1 SG	C Band SAR	LEO
					C, L Band SAR + Rad	LEO
Earth Optical Imaging	Sentinel-2 A Sentinel-2 B	Sentinel-2 C Sentinel-2 D	Sentinel-8 Sentinel-10	Sentinel-2 SG	Multispectral	LEO
					TIRI	LEO
				Hypespectral	LEO	
				MSI+TIRI SG Hyper-spectral SG OLCI+SLSTR SG	LEO	
			Sentinel-3 SG	OPT	GEO	
Earth Topography by Altimetry	Sentinel-3 A Sentinel-3 B	Sentinel-3 C Sentinel-3 D	Sent.-9 TOPO	Sentinel-6 SG Sentinel-9 SG	C/Ku Radar Altimeter	LEO
	Sentinel-6 A Sentinel-6 B	Sentinel-6 C Sentinel-6 D			VIS-SWIR-TIR radiometer	LEO
					C/Ku Altimeter	LEO
					GNSS Radio-Occultation	LEO
					Altimeter ICE/SNOW	HEO
			Sentinel-9 HEO	Altimeter LEO SG	LEO	
				Altimeter ref.SG	LEO	
Earth Atmosphere by Spectroscopy	Sentinel-4 A Sentinel-4 B	Sentinel-5 C	Sentinel-7	Sentinel-9 HEO SG Sentinel-4 SG Sentinel-5 SG Sentinel-7 SG	Altimeter ICE SG	HEO
	Sentinel-5 A Sentinel-5 B				UVN Spectrometer GEO	GEO
	Sentinel-5 P				Infrared Sounder GEO	GEO
					UVN Spectrometer LEO	LEO
					Infrared Sounder LEO	LEO
					Spectrometer LEO	LEO
					CO2 Spectrometer	LEO
					HEO Spectrometer SG	GEO
				GEO Spectrometer SG	LEO	
				LEO Spectrometer SG	LEO	
				CO2 Spectrometer SG	HEO	

# Outlook: Earth Explorers (EE)



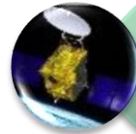
## Core Missions



**GOCE** (2009–13) studying Earth's gravity field



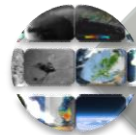
**ADM-Aeolus** (2017) studying global winds



**EarthCARE** (2018) studying Earth's clouds, aerosols and radiation (ESA/JAXA)



**Biomass** (2021) studying Earth's carbon cycle



**EE 10 Call** to be released end of 2017

## Opportunity/Fast Track



**SMOS** (2009– ) studying Earth's water cycle



**CryoSat-2** (2010– ) studying Earth's ice cover



**Swarm** (2013– ) three satellites studying Earth's magnetic field

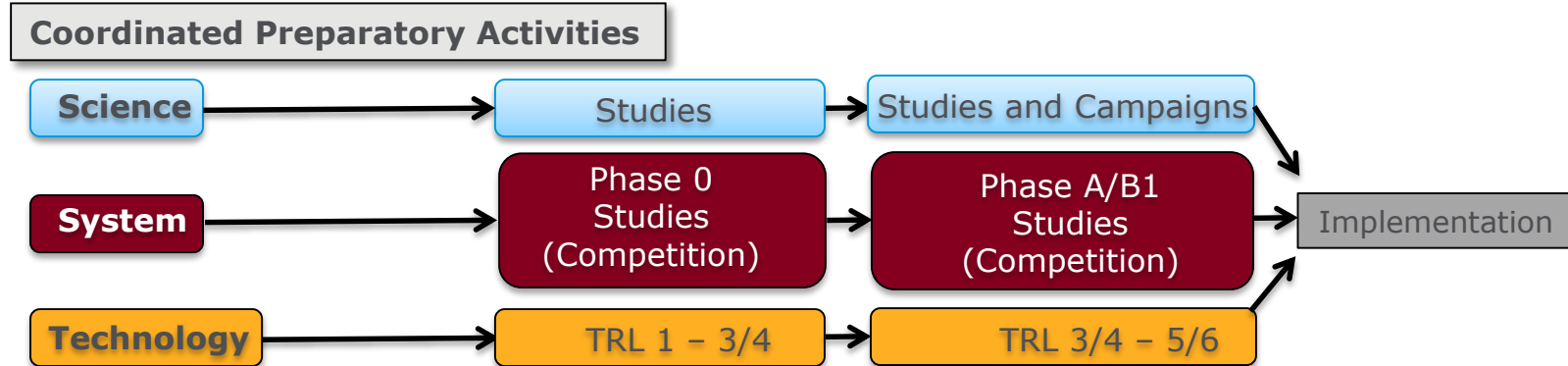
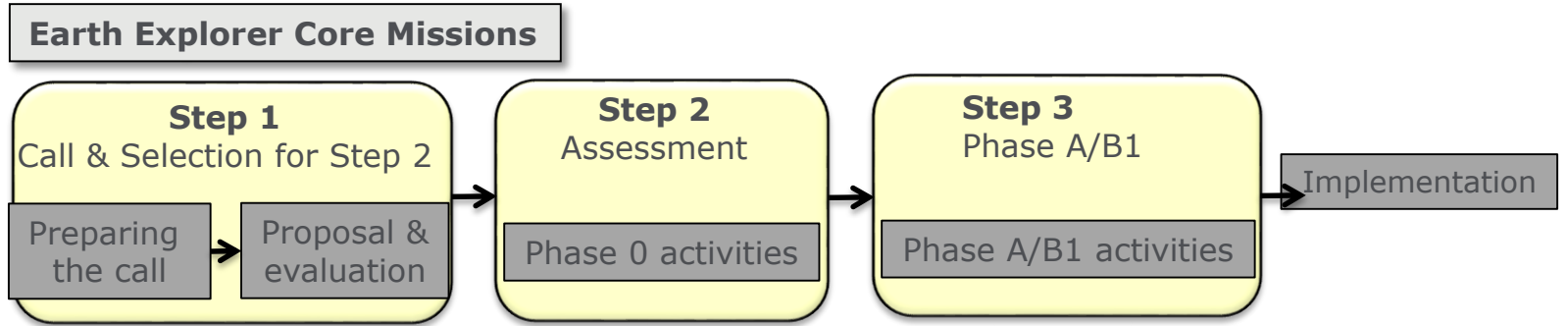


**FLEX** (2022) studying photosynthesis

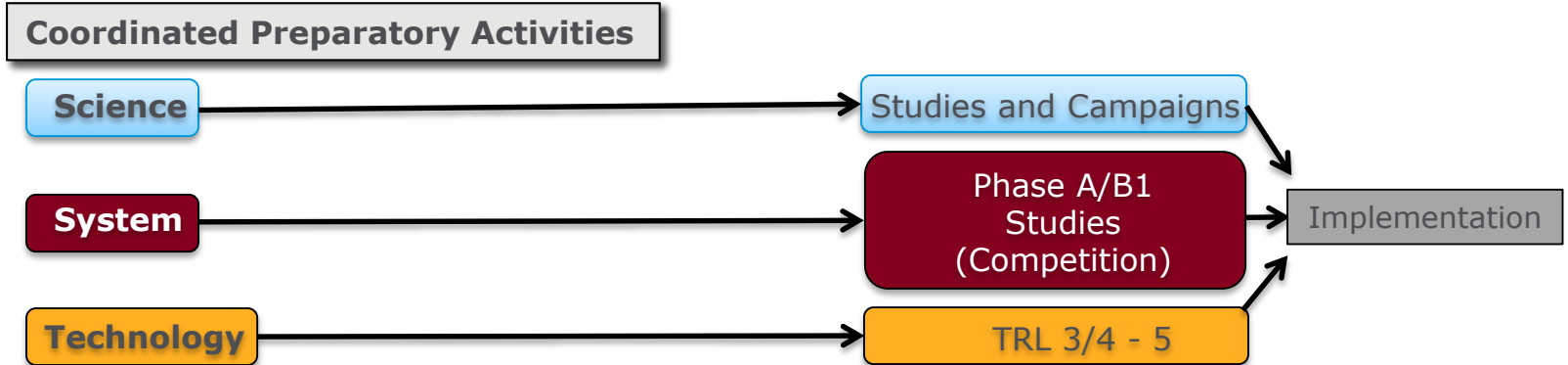
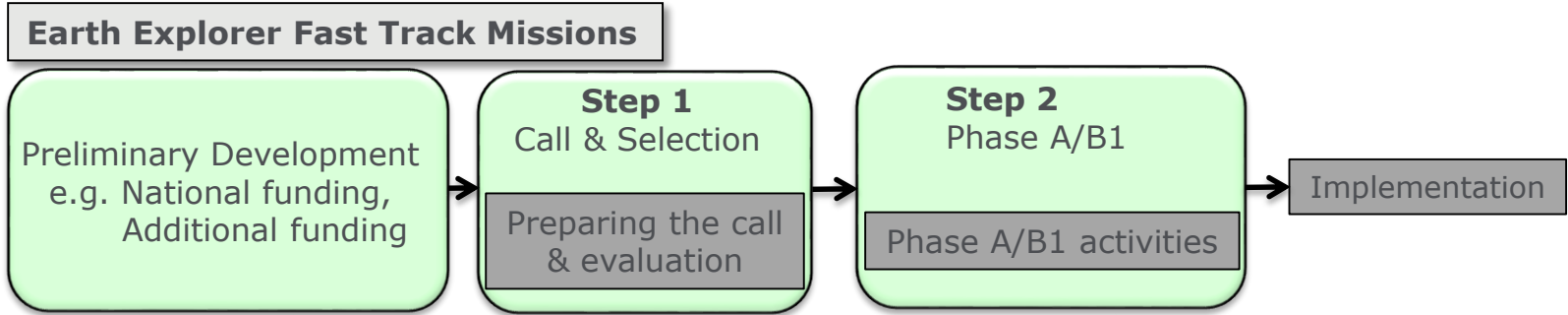


**EE 9** proposals expected for 1st Jun. '17, selection by PB-EO is 1. Nov. '17

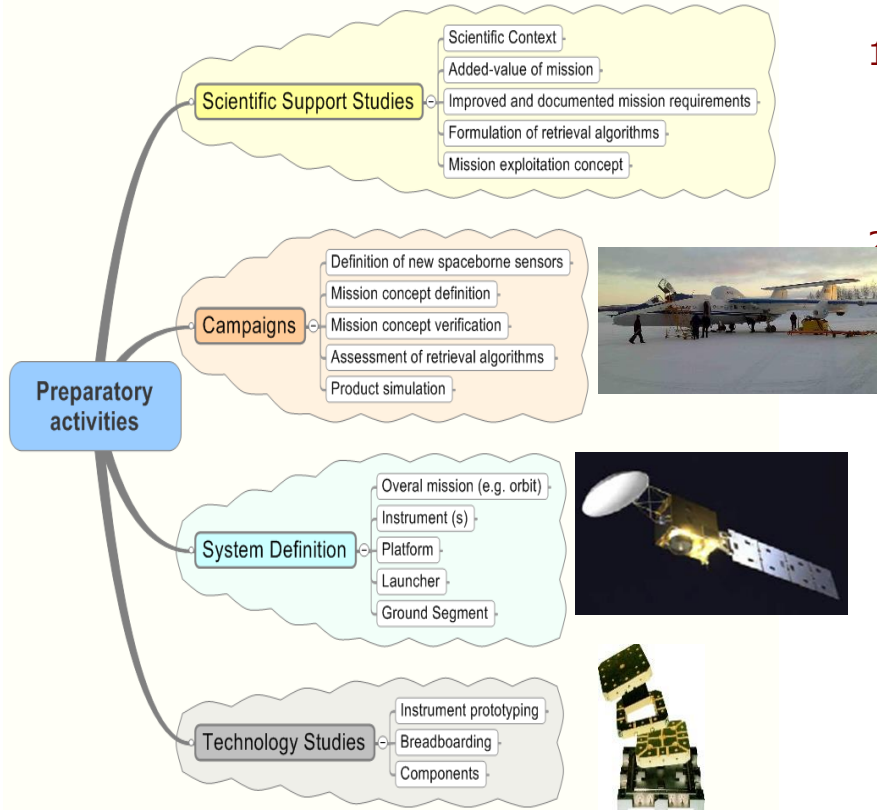
# EE Core mission activities vs preparatory activities



# EE Fast track mission activities vs preparatory activities



# What is a preparatory activity?



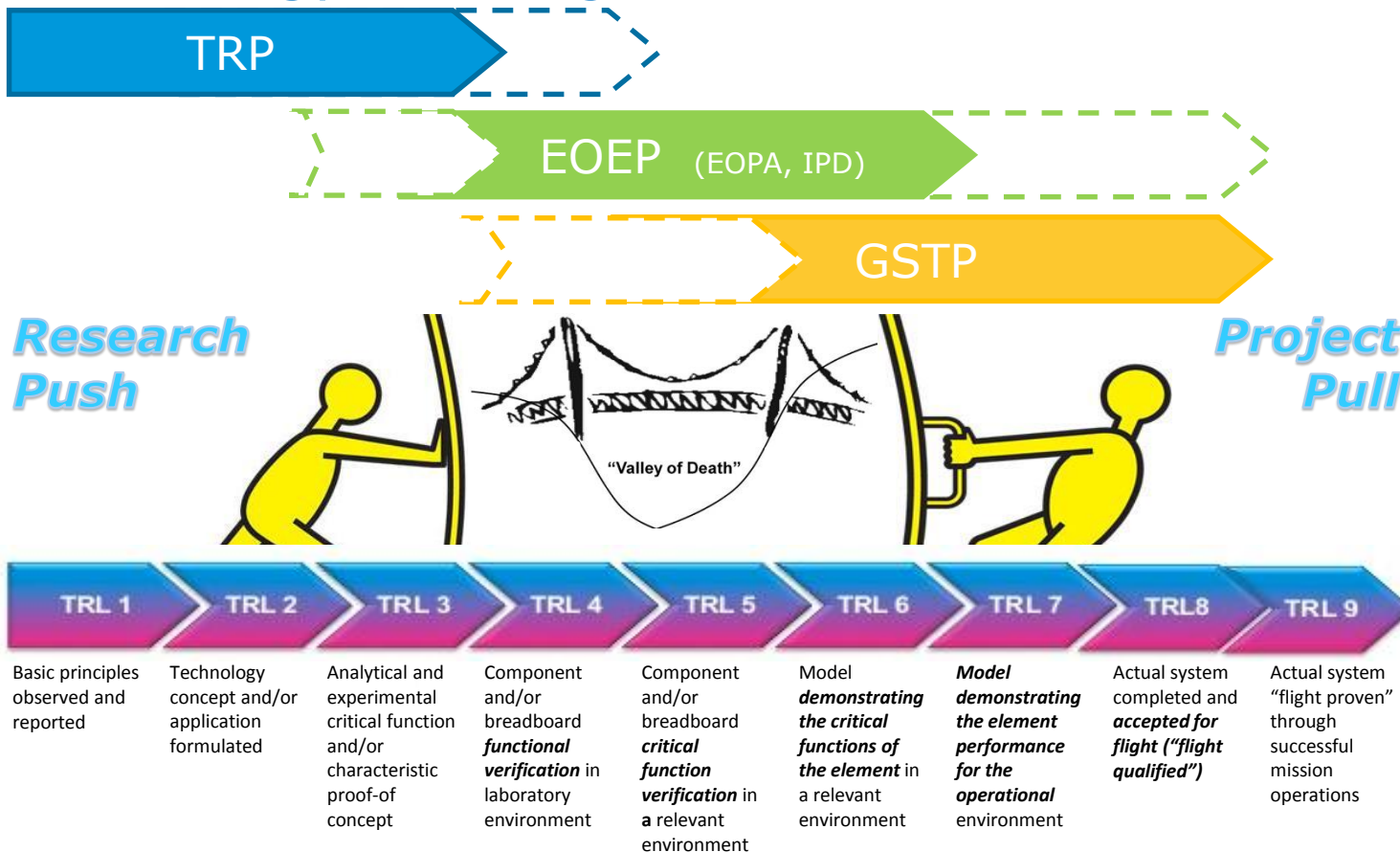
1. EOEP preparatory activities include all necessary activities to define and evaluate future EO space borne missions (Earth Explorer, Copernicus, meteorological,..)

2. Driving elements include:

- Scientific challenges summarised in “The Changing Earth”, SP-1304
- Associated observation, mission and technology requirements
- Mission preparation through Phase-0 (Pre-feasibility) and Phase-A/B1 (Feasibility)
- Foster new ideas, cooperation opportunities and prepare technologies, also for European independent capabilities
- ESAC recommendations



# EOP technology founding scheme vs TRL level



# EECM preparatory activities: BIOMASS example

## Example: BIOMASS Core Mission Phase A (now in Phase B)

### System

- BIOMASS Phase A System Study (x2), addressing:
  - Space segment : payload, platform
  - Mission analysis and operations
  - Launcher
  - Ground segment
  - Critical technologies
  - Programmatics

### Technology

- Large P-Band SAR antennas critical breadboard (x2)
- Very Large P-Band Antennas performance verification methodology & Facilities
- P-Band HPA technology assessment
- Very large space antenna aperture – demo model
- P-Band Reflector antenna Feed elements
- P-band ice sounding radar demo development
- P-band passive sub-array development
- Very large space antenna aperture architecture trade-off (x2)
- SSPA breadboard (incl. circulator/switch, power divider and calibration coupler) (x2)
- Study of P-Band transponder with ionospheric correction capabilities (x2)

### Science and Campaigns

- Development of algorithms for forest biomass retrieval
- Study of ionospheric disturbance mitigation schemes
- Assessment of the BIOMASS retrieval error on flux
- P-Band SAR wave interaction and information retrieval
- Analysis of BIOMASS secondary objectives

e.g.

- TropiSAR campaign (completed)
- TropiScat campaign (on-going)
- BioSAR 2 campaign (completed)
- BioSAR 3 campaign (completed)

### End-to-end Performance Evaluation and System Support

- BIOMASS End-to-End Mission Performance Simulator
- OpenSF end-to-end (E2E) simulator framework infrastructure
- Modern attitude control of EO satellites with large flexible elements (x2)

● = EOPA

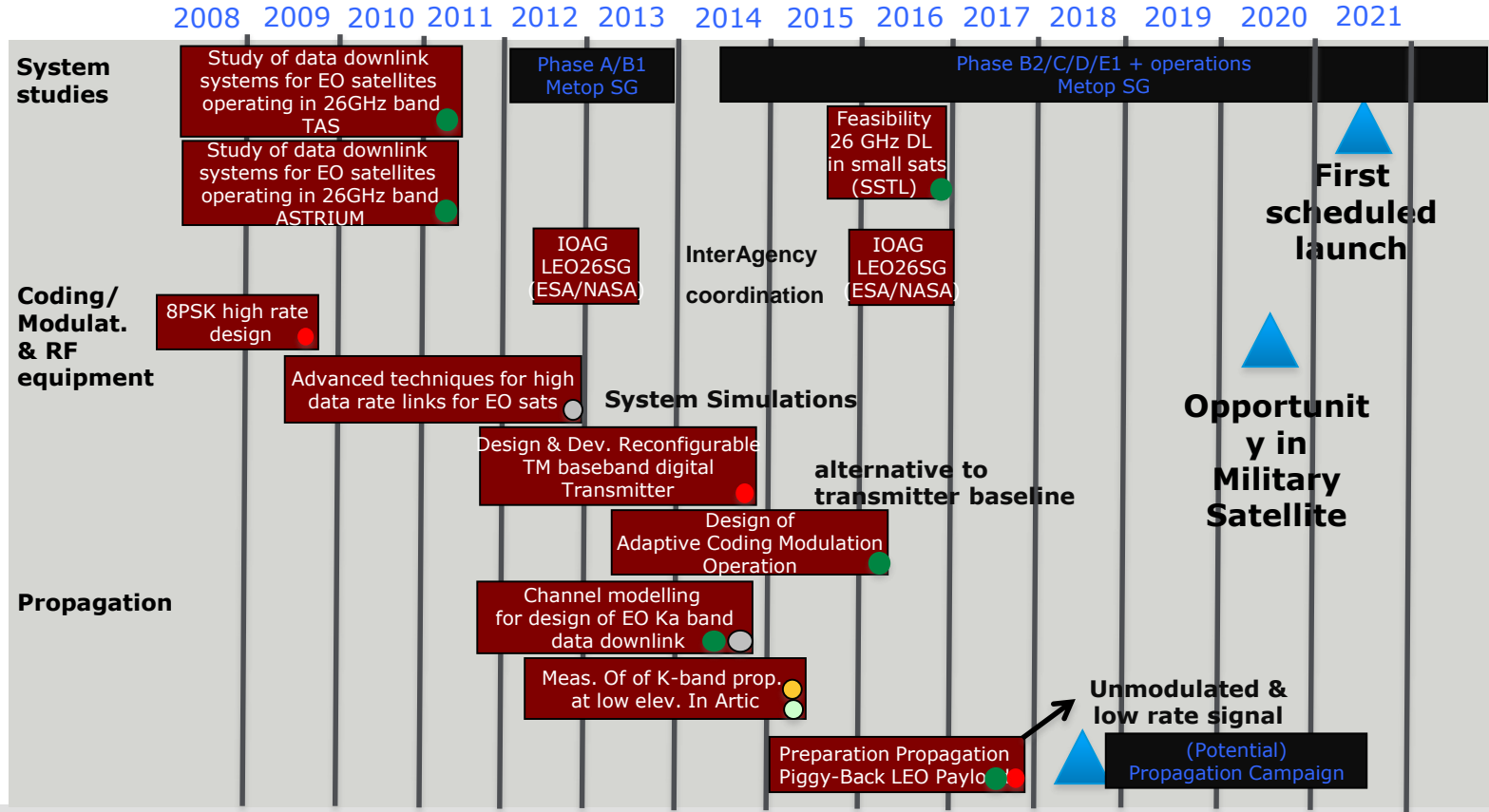
● = IPD (also EOP)

● = TRP / GSTP

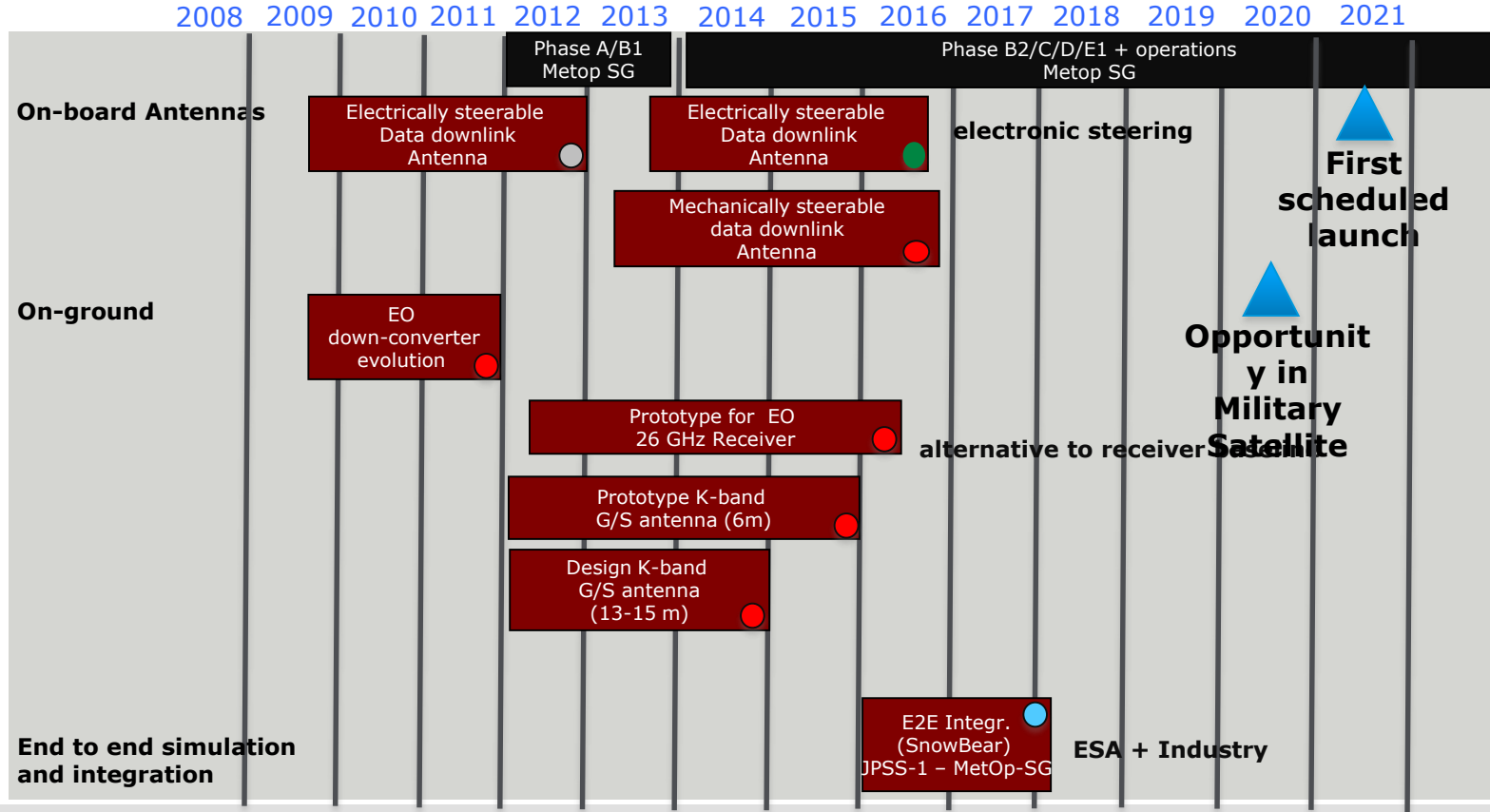
● = GSP

Slide 18

# Example : Overview of 26GHz Downlink Development

# Example : Overview of 26GHz Downlink Development

# Achievements - TRP 4000101855 – Swath Precipitation Radar Instrument



**Airbus Defense & Space (UK)**

**Duration – 24 months**

## Background and justification

Numerical weather prediction models are continuously updated as new observations become available. Scatterometer derived winds are routinely assimilated, but they provide no information on the vertical profile of the wind. This is vital for improved weather forecasting and climate change predictions.

## Objective

To develop a radar instrument to provide wide coverage of 3D vectors of tropospheric winds and precipitation which would operate in cloudy regions and hence complement ADM-Aeolus

## Achievements and status

The original activity was re-aligned to focus more on the wind vector extraction than on precipitation. The result is a conically scanning radar at 94GHz which emits pulse polarization pairs in order to cover a swath of 800km with 50km horizontal and 1km vertical resolution. It is capable of measuring wind speeds up to 160m/s with an accuracy <2m/s. The instrument has reached the stage of an advanced concept. Renamed Wind Retrieval Radar Nephoscope – WIVERN

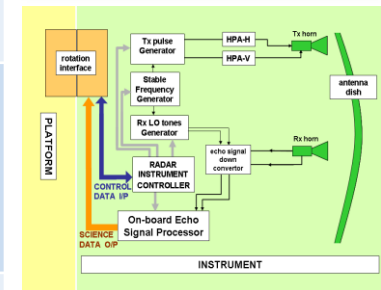
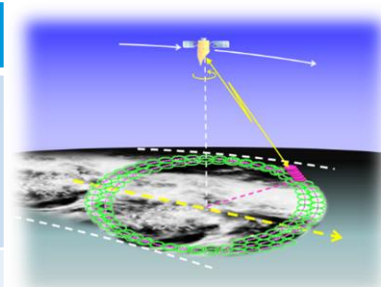
## Benefits (vs expectation)

An instrument has been conceived which would add significantly to scientific understanding of weather and climate change

## Next steps

Under the IPD budget, ground tests are being carried out at Chilbolton (UK) and airborne tests over the Great Lakes in Canada to prove the concept.

Initial TRL 1	Current 2	Final 3	and/or Target TRL 5
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European Space Agency

# Achievements - T117-309MM + T116-044MM

## High-Stability Laser with Fiber Amplifier + Laser Stabilization Unit for Interferometric Earth Gravity Measurements - 3/y MJ

SpaceTech (D), Fraunhofer ILT (D), ADS (D), NPL (UK)

Duration: 36 months

### **Background and justification:**

Laser technology development for a Next Generation Gravity Mission (NGGM) as GOCE follow-on. The NGGM measurement principle is based on laser interferometry between two satellites flying in a LEO formation with up to 200 km distance. Laser frequency stability requirements are one order of magnitude more challenging than the current specs for eLISA.

### **Objective:**

Develop an elegant breadboard of a high-stability laser consisting of a master oscillator (MO), fibre-based power amplifier (FPA) and an optical cavity (OC) for frequency stabilisation. Demonstration of a space-worthy design for all the subunits (representative environment and interfaces, limited environmental testing on component and subunit level).

### **Achievements and status:**

Completed with full achievement of all objectives in terms of performance, thermal and mechanical modelling as well as radiation and thermal testing.

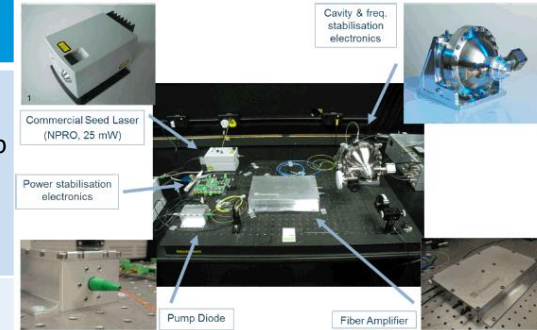
### **Benefits (vs expectation):**

Space-worthy hardware (HSL-EBB) ready for further environmental testing.

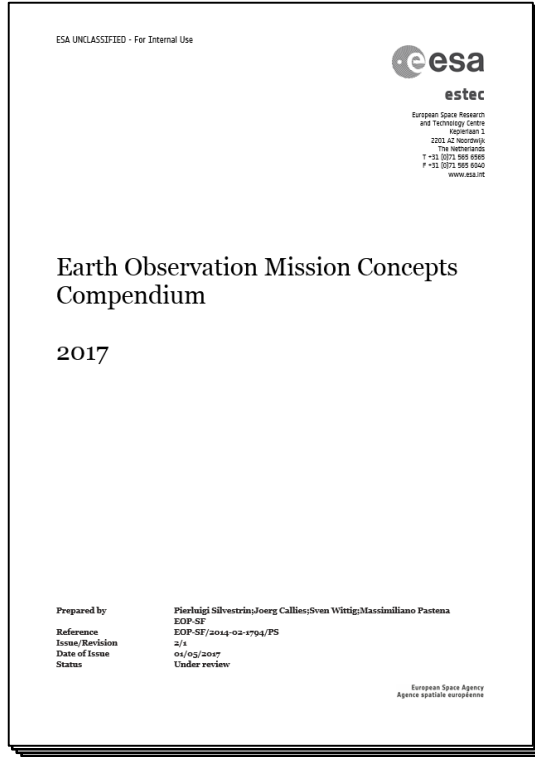
### **Next steps:**

Raise TRL of FPA and OC to TRL 5 (subunit) / TRL 7 (components) within GSTP

Initial TRL	3	Current	Final	4	Target TRL	4
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# Earth Observation Mission Concepts Compendium 2017



## Overview of over 40 EO mission concepts

- From P-band to UV
- Constellations and Companions
- Concepts vary very strongly in their definition (from pre-phase 0 to full phase-A)  
(e.g. CDF studies to EE7 candidates like Premier, CoreH2O)
- System Concepts & Technology funded by:
  - EOEP lines (EOPA, EWD, IPD, STSE)
  - and/or corporate programmes (TRP/GSTP)





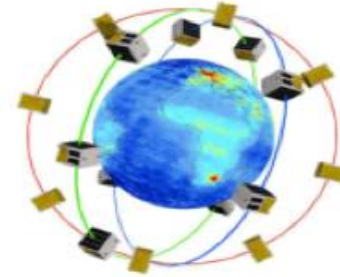
# Each Concept has a Single Page Summary

Example →

Key concept parameters

## 12 OMNISAT-HAPI CONSTELLATION

<b>Science Domain</b>	Atmosphere
<b>Instrument Type</b>	Multispectral Imager
<b>Wavebands</b>	VIS
<b>Spat. Resolution</b>	400 - 600 m
<b>Swath</b>	200 km
<b>Orbit / Lifetime</b>	SSO / 4 years
<b>LDTN</b>	08:00 / 12:00 / 16:00
<b>Revisit</b>	1 hour
<b>Coverage</b>	Global
<b>Satellites</b>	24 x (constellation)



**Wet Mass**  
38 kg ( ✓ Vega )

**Peak Power**  
60 W

**Downlink**  
5 Mbit/s

Objectives

### Mission Objective:

Measurement of tropospheric nitrogen dioxide (NO<sub>2</sub>) content with high temporal resolution to issue NO<sub>2</sub>-related health alerts in near real-time for city targets.

Origin

### Concept Origin:

In 2015 the OmniSat-HAPI (High-resolution Anthropogenic Pollution Imager) Nanosat constellation mission was selected as one of two winners of the ESA SysNova R&D Studies Competition for Innovation.

The mission concept was then further reviewed during a CDF study at ESA [RD-65-1].

# Concept in Detail (example)

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## Science and Application

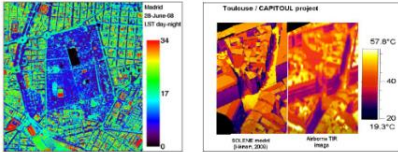


Figure 11-3: Left: Difference between land surface temperature at day and night over Madrid (Scherino et al., 2009). Right: Comparison of canopy surface temperature over Toulouse (Heron, 2008)

The TIREX mission studies the land surface temperature. The land surface is coupled to the climate in the Earth System through a wide range of biogeophysical and biogeochemical processes including the energy budget, the water cycle, and the carbon cycle. Land surface temperature (LST) is an integrated indicator to the coupled energy-water-carbon dynamics.

**Why**

LST is highly useful to better constrain modeling these land surface models and the nature of landscape heterogeneity. Tropical forest areas which are high and in the range of spatial and temporal observations of LST in support of water and carbon cycle. Several attempts have been made to exploit the daily MODIS and MSG systems to monitor and monitor vegetation/climate system and for agricultural and hydrological applications, however the methods are not robust enough because of the mismatch between the spatial resolution and the local variability.

The proposed mission is a thermal imager which combines a high spatial resolution (50 m. minimum) with a high (1 day) revisit time; a combination that could significantly improve modeling capabilities in surface energy water and carbon cycles for land applications. TIREX will provide simultaneous thermal infrared images and solar spectrum images at constant viewing angles and constant local time. The constant viewing angle allows reducing directional effects and gives far better accuracy for pixel-to-pixel or day-to-day comparisons. Key products (at 50 m, 1 day revisit) of TIREX are top of atmosphere reflectance and surface reflectance (from VNIR), brightness temperature and land surface temperature (from TIR), and derivative biophysical variables of emissivity, albedo, vegetation cover, and LAI. The primary objective of TIREX is to monitor the biogeochemical cycles (water and carbon dioxide in particular) thus contributing to applications in the realms of agriculture, forestry and hydrology related to climate change. A second mission objective focuses on urban environments and thirdly the focus is on the coastal (water) areas. In addition, the mission allows studying various high temperature events (e.g., volcanic eruptions, forest fires, coal fires etc.).

Page 57/903  
Earth Observation Mission Concepts Compendium  
Issue Date 26/10/2018, Ref EOP-SF/2014-02-1794/PS

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## Sensing Concept

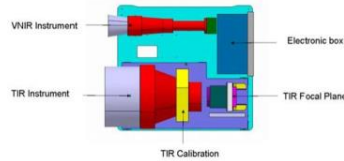


Figure 11-4 CAD view of TIREX Instrument

For TIREX the mission concept envisages two identical micro-satellites that will fly on the same two-day repeat sun-synchronous orbit at 720 km altitude separated by 180°. The satellite performs roll and pitch manoeuvres ( $\pm 35^\circ$ ) to offer daily access to any target on the Earth. Each satellite carries a Thermal InfraRed (TIR) multispectral imager and a Visible Near InfraRed (VNIR) imager. The VNIR channels cover a swath of 25 km with a resolution of 20  $\mu\text{m}$  with a width of 20 to 40 nm in the VNIR and 0.5  $\mu\text{m}$  in the TIR channels is about 100  $\mu\text{m}$ . The delta Temperature (NETD) of either squares with size of 25  $\mu\text{m}$  is 0.1 K. The TIR instrument operates in pushbroom mode. The focal plane uses off-the-shelf uncooled micro-bolometers. The instrument combines the "supermode" concept with Time Delay Integration (TDI) function to improve SNR performance. The "supermode" operation is achieved by tilting the detector array by 45° with respect to the along-track direction. The spectral channels are in-field separated, their selection being realised by an interference filter in front of the detector. The TIR channels are on-board calibrated against blackbodies and deep space to an accuracy of 1 K. A refractive telescope is proposed with a pupil size of 270 mm. The description of the VNIR instrument is missing.

**How**

In order to enable an accurate use of time series of images, multi-temporal registration to better than 0.3 pixel is required. Automatic registration with Ground Control Points (GCP) is proposed, using algorithms developed for the Pleiades and Sentinel-2 missions. The TIREX platform is based on the latest version of the standard Myriade platform, enhanced with more capable reaction wheels for agility purpose. The mission duration is nominally five years. Two scenarios are considered for the launch: one Soyuz launch with two TIREX satellites as auxiliary payload in ASAP launch configuration and two dedicated launches by VEGA.

Page 57/903  
Earth Observation Mission Concepts Compendium  
Issue Date 26/10/2018, Ref EOP-SF/2014-02-1794/PS

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## 8.3 Critical Technologies

Table 8-3 GEO-OCULUS TRL Overview

Item	Current maturity	Roadmap to Phase C/D
$\varnothing$ 1.5 m imaging telescope	TRL 4	Manufacturing process validation in on-going GSTP
10k x 10k CMOS imaging array	TRL 2/3	Array sizes are beyond state-of-the-art, pre-development is required to validate foundry process & CMOS stitching
2k x 2k SWIR/MWIR hybrid array	TRL 2/3	Back-illuminated CMOS to be validated
0.8k x 0.8k TIR hybrid array	TRL 2/3	Back-illuminated CMOS to be validated
Cryocoolers	TRL 5/6	Microfabrication mastering for Pulse Tube technology and/or increased Stirling cooler capacity for TIR
QVD Sun diffuser for UV-VNIR	TRL 7/8	Will be validated in orbit in 2009 by COMS/GOCI
On-board image processing for:	TRL 2	Prototyping of image processing algorithms & software

**Effort & Risk**

Due to resource constraints, the proposed technology requirements were not fully progressed in the context of Geo-Oculus. However, some technology items were progressed for other missions.

In a further GSP study "Towards 1-m from GEO" the option of very high resolution observations were studied for which many technology breakthrough areas were defined (e.g. deployable solid optics).

For resolutions below 1 m from a geostationary orbit synthetic aperture optics would allow even better resolution but the satellite and the instrument become more complex. A Synthetic aperture study was conducted in 2008 (Applications for Aperture Synthesis Techniques for imaging in Earth observation and Science).

Page 57/903  
Earth Observation Mission Concepts Compendium  
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