

A low cost upper atmosphere sounder (LOCUS) mission

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So what is LOCUS?



A breakthrough concept multi-terahertz remote sounder

Compact payload to be flown on a 'standard' small satellite that will:

- Measure key species in the upper atmosphere, i.e. the mesosphere and lower thermosphere (MLT);
- Increase understanding of natural and anthropogenic effect on climate change;
- Allow study of the 'gateway' between the Earth's atmosphere and near space environment.



Small satellite sounder from LEO



- Terahertz (THz) frequencies (sub-mm-waves) penetrate dielectric media opaque at most other shorter wavelengths;
- Can detect and characterize molecular species through obscurants and located in a relatively low temperature environment;
- Offers higher spatial resolution than microwave/mm-wave range;
- Allows remote sounding of atmospheric constituents related to climate change on a local or global basis;
- Same technique provides information on the interstellar media, e.g. regions of star formation.



LOCUS science achieved through:

- Tracing O, OH, NO, CO, O₃, H₂O, HO₂, O₂ spectral emission signatures globally and from low Earth orbit (LEO);
- Using a limb sounding technique with cold space as a background to achieve height distribution;
- Provision of ultra-high spectral resolution (1MHz);
- Accurate spatial sampling with ~2km footprint at tangent heights from ~ 55km to 150km.

But, this requires terahertz instrumentation:

Species	Transition Frequency (THz)
0	4.745
OH	3.544
HO ₂	3.543
	3.544
NO	1.153
	1.153
СО	1.152
02	0.773

And therein lies one challenge...





IMPATT – Impact Ionization **Avalanche Transit-Time** diode **HG** – Harmonic Generation **RTD** – Resonant-Tunnelling Diode **TPO – THz Parametric Oscillator** PCS – Photoconductive Switch QCL – Quantum Cascade Laser

M. Tonouchi, Nature Photonics, 1, 97 (2007)

To be based on heterodyne receiver technology, which converts THz input signal to a lower intermediate frequency (IF) – typically GHz;

Provides low noise and high spectral resolving power - order >>10⁴; Key front-end components are the mixer and THz local oscillator (LO).



The heterodyne approach



Optimum system performance requires:

- Efficient signal frequency translation, i.e. low conversion loss;
- Minimal added system electrical noise;
- Provision of adequate THz LO source power.





e.g. RAL & STAR Dundee 350GHz Heterodyne Radiometer (previous CEOI funding)

The LOCUS payload concept









- Highly integrated multi-channel THz radiometer system;
- Four separate bands identified that accommodate the required spectral windows;
- Schottky semiconductor diode mixer technology;
- Quantum Cascade Laser used a LOs for 1 and 2, harmonic up-conversion for 3 and 4;
- Fast Fourier Transform digital spectrometers provide 1MHz spectral resolution;
- Single primary ~ 40cm diameter and miniature coolers 100K operational goal;
- UK sourced technology with critical elements support by the CEOI and NERC.

Electronic behaviour





Peak performance corresponds to efficient injection of current

Device dimensions are typically 1 mm \times 150 μm \times 10 μm



Quantum Cascade Lasers (2)

- 1 W peak power is possible, and 10s of mW continuouswave power, *but cooled*;
- QCLs have an intrinsically narrow linewidth (<20 kHz);
- Precise frequencies can be defined using periodic gratings defined into the ridge waveguides;
- Operation has been demonstrated over the 1 – 5 THz frequencies;
- Radiation hardness has been demonstrated.



2% duty cycle; ;asing up to 1.01 W (peak) at ~ 3.4 THz; > 400 μW at 77 K, T_{max} = 118 K: L. Li *et al*, *Electronics Letters* 50, 309 (2014).

LOCUS Core Technology





In-orbit demonstration – satellite concept



Objective: Prove core payload and platform technology in space:

- Polar sun synchronous orbit;
- Perform global species measurement;
- Novel approach to scene scanning via spacecraft nodding;
- Cold-space view and on-board c300K target provide payload calibration;
- Approx. <u>total</u> spacecraft volume, mass & power: 1m³, 150kg, 70W.
 - Compare with NASA AURA @ 43m³, 3tonne, 4kW
 & MLS: ~8m³, 500kg, 550W);
- IOD mission lifetime ~ 2 years, tbc.

Mission Concept Development Plan UNIVERSITY OF LEEDS





ESA In-orbit Demonstration Study Programme (SSTL PI):

- Science refinement;
- Payload concept definition;
- Spacecraft concept definition;
- Mission plan and cost estimate.

NERC Critical Component Development (RAL PI):

- QCL development and waveguide demonstration;
- THz Schottky diode development;
- Integrated QCL & Schottky proof of concept.

CEOI-ST Critical Payload Development (Leeds PI):

- 1.1 THz (Band 3) full development including:
 o Mixer, LO and spectrometer;
- QCL frequency stabilisation.



- THz remote sounding provides important information in relation to the Earth's climate evolution and its monitoring;
- The THz detection method depends upon the nature of the defined science return;
- Where the science requires high spectral resolution and high sensitivity, THz heterodyne detection is the instrumentation of choice;
- In the 1 to 5 THz frequency range, novel heterodyne instrumentation is being conceived and developed that will allow novel scientific study;
- A UK initiated and presently majority UK funded instrument, LOCUS, is being developed to study the relatively unexplored supra-THz spectral range.

The LOCUS Team Members





Thank you for your attention