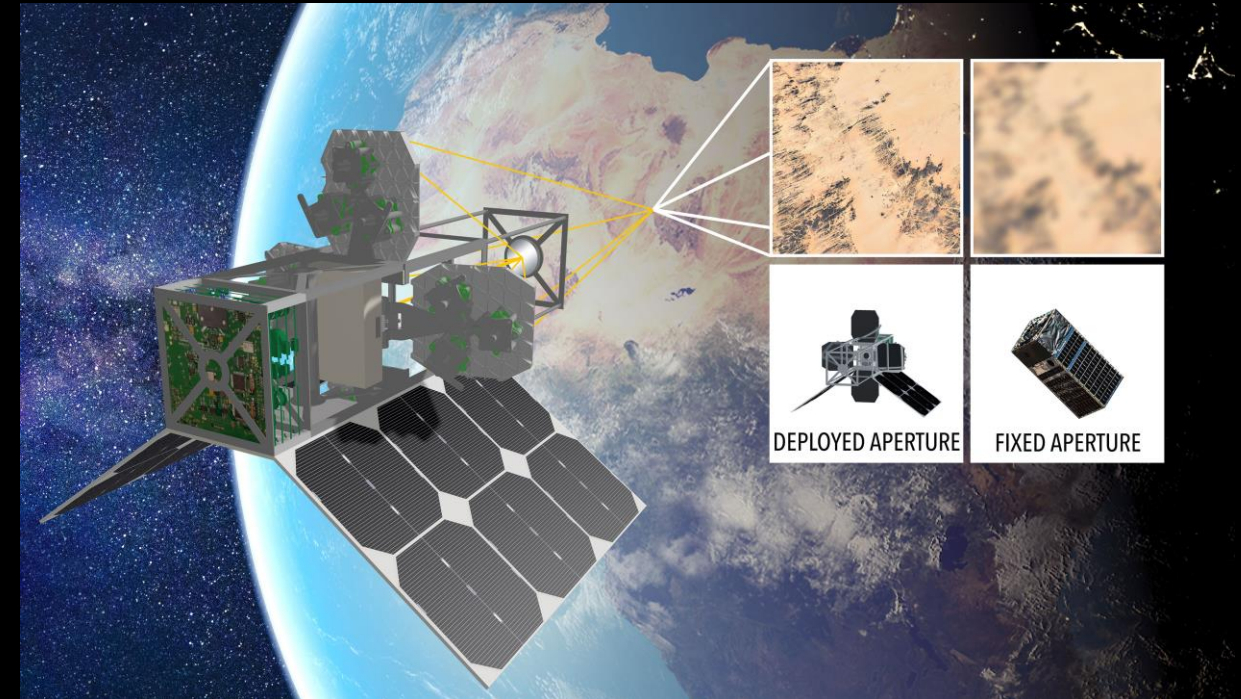


A-DOT

A small deployable telescope for EO and astronomy

Noah Schwartz, Douglas Harvey, Éamonn Harvey, Katie Morris, Junyi Zhou, Maxime Dumont, Jean-François Sauvage, Carlos Correia, Jay Stephan, Samuel Pinilla Sanchez, Siu-lun Yeung



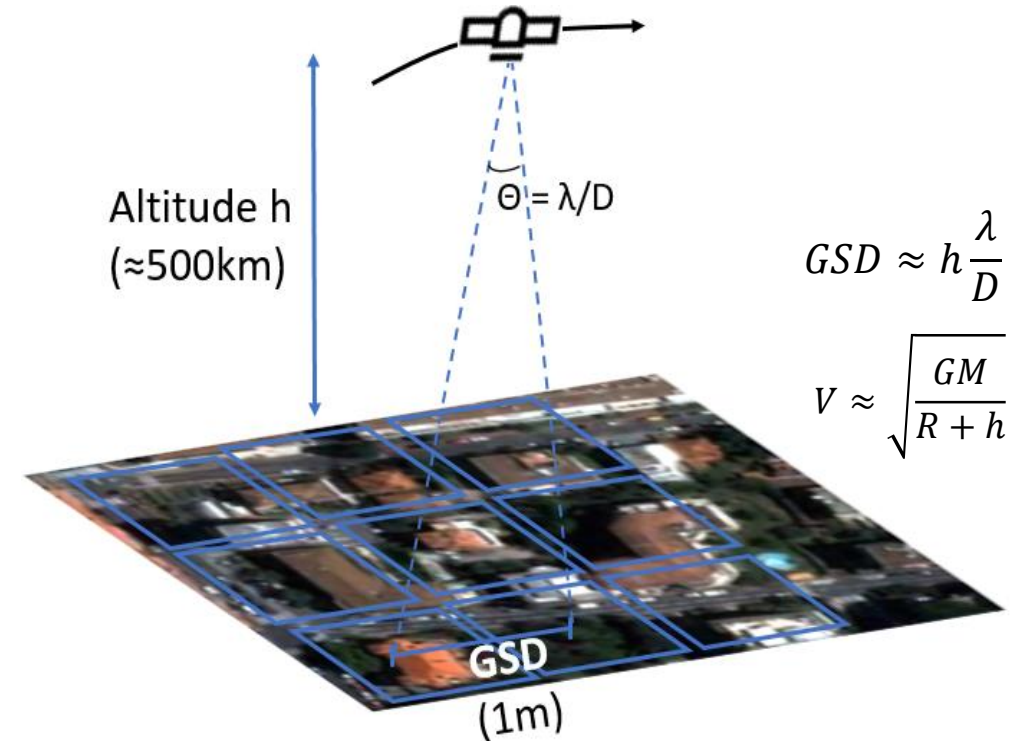
High spatial resolution in the optical

New space era (circa 2000s) still limited by *old* physics (circa 1660s)

Resolution limited by diffraction

Revisit period limited by orbit (& cost)

Pupil of 30cm		GSD = 1m
λ (μm)	GSD	D
0.5	1 m	0.3 m
2.5	5 m	1.5 m
5	10 m	3 m



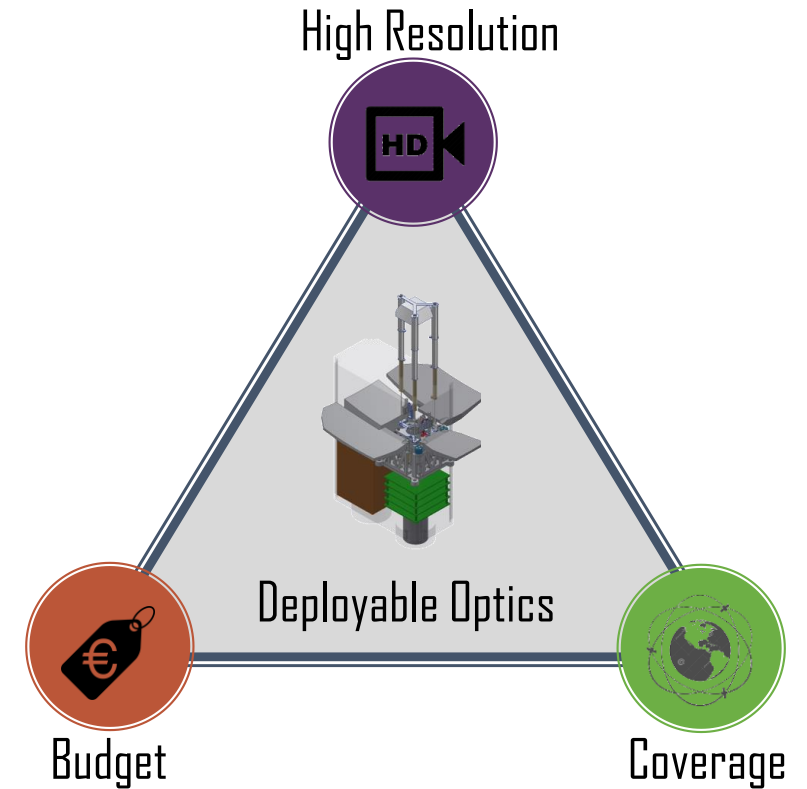
High resolution optical imager

- 1 m GSD from LEO & 5 km swath
- $\lambda = 500\text{-}800$ nm (optical wavelength)
- Small and lightweight platform

➔ **Deployable optics**

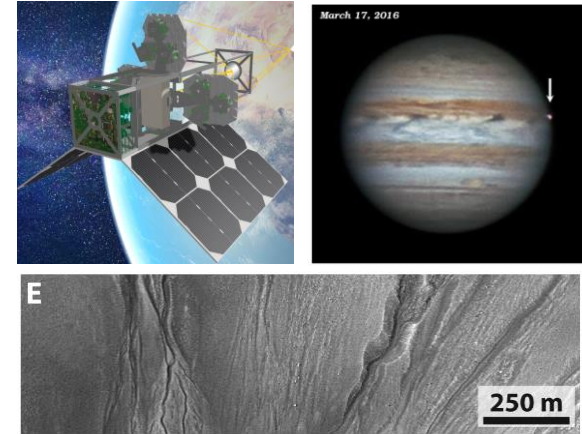
Active Optical Deployable Telescope

- 30 cm deployable aperture
- 6U CubeSat platform
- Compared to non-deployable 6U
 - x3 resolution, x7 flux, x10 mass reduction



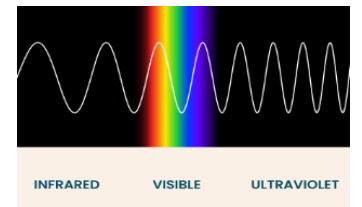
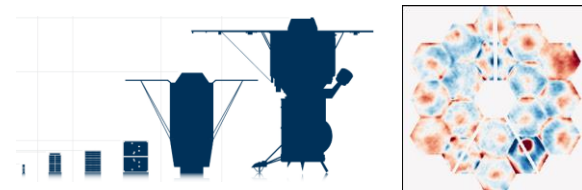
Enabling (very) high-resolution images

- On small platforms
- Potential for high-revisit rates
 - EO: Earth climate monitoring and protection, defence and security, natural disasters monitoring and management plans...
 - Astronomy



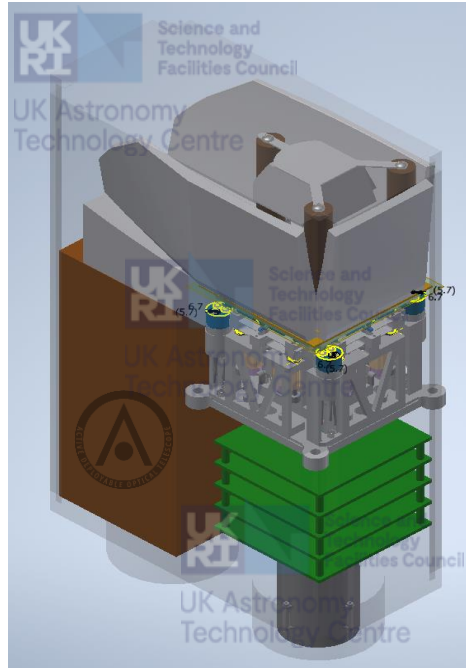
Building technology to allow affordable HR imaging

- Wavefront control is at the heart of ADOT
- Potential for missions with large aperture telescopes
- Potential for longer wavelengths & different applications (e.g. LiDAR)
- We hope that the UK can get a lead in this and place ourselves at the heart of these missions

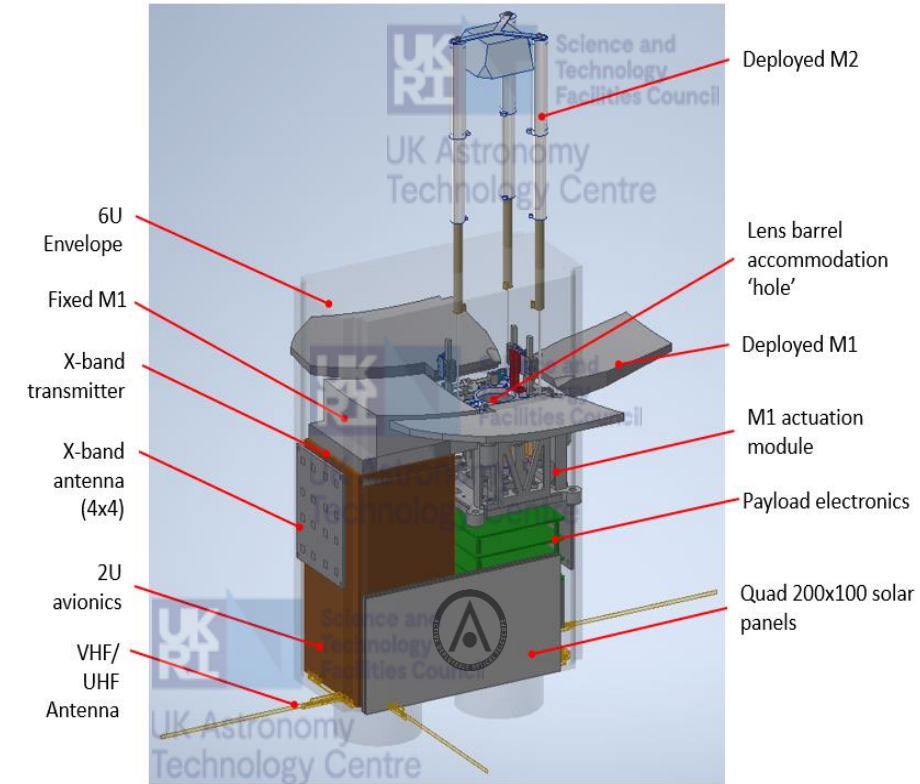
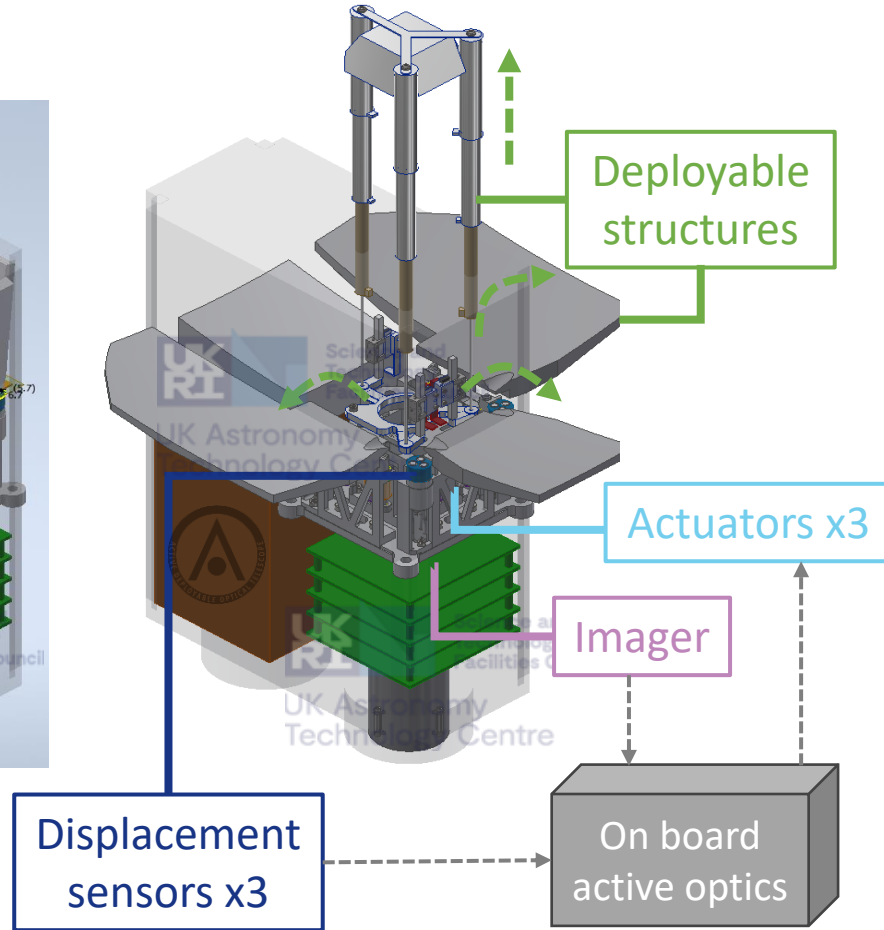


Active Deployable Optical Telescope

Folded



Deployed



Key technical challenges

Image quality criterion

- Maintain diffraction-limited system
- **70 nm RMS** residual error

Opto-mechanical design

- Thigh volume allocation
- Precise control of optics
 - Deploy $\lesssim 10 \mu\text{m}$ of nominal position
 - Control within $\lesssim 10 \text{ nm}$

Active optics

- Maintain alignment of telescope
- Automatically during operations

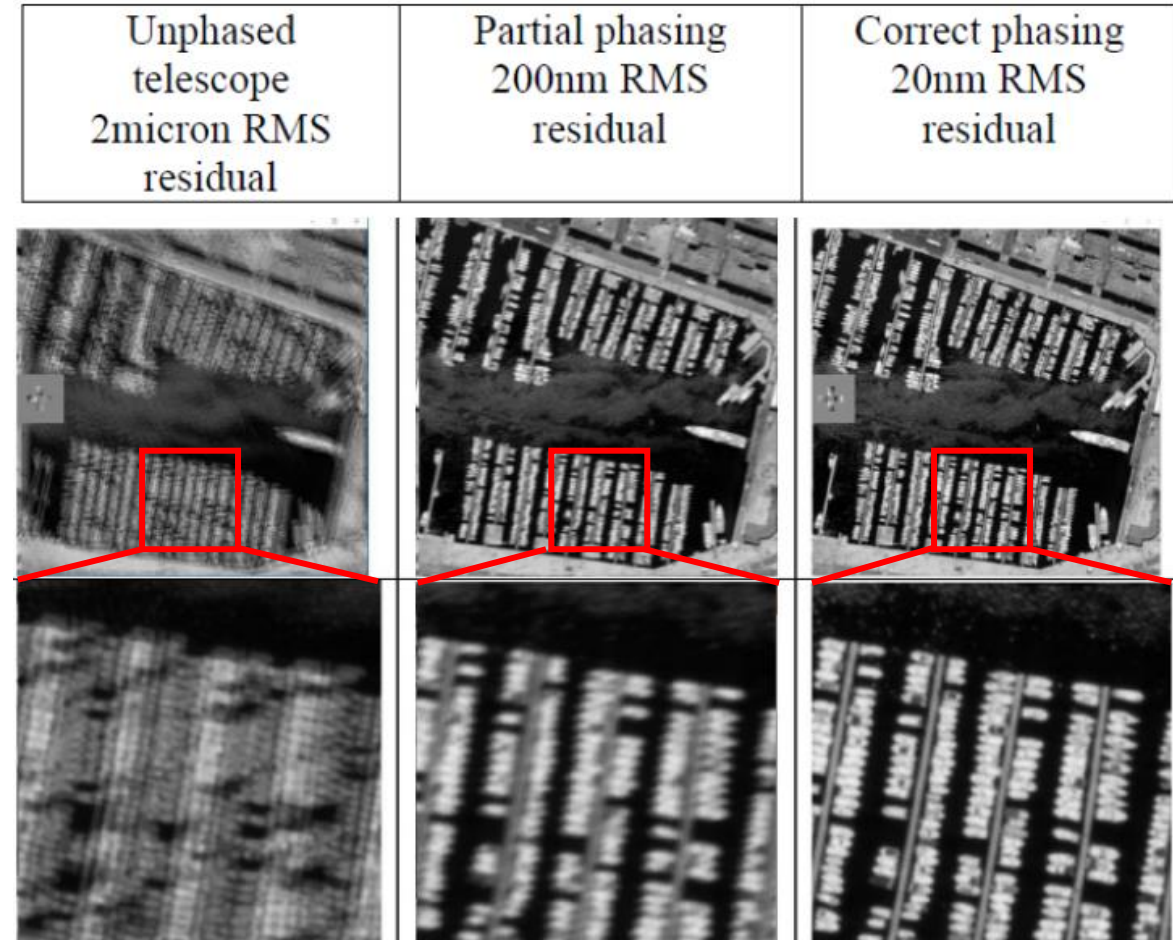
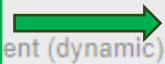


Image Quality error budget

Category	Definition	Error Source	Comment	Error #	WFE budget
Intrinsic optical errors	Quality of optics	Manufacturing	Low-order residual manufacturing error of the optics (mostly M1 and M2)	1	20
		Polishing	Polishing errors, mainly high-orders	2	30
	Initial alignment (on the ground)	Integration / alignment	The optical residual of the optics after alignment of the system in laboratory, excl. segment phasing residuals (e.g., lenses alignment, M1 positioning accuracy)	3	30
	Ground to space degradation (static or one-time events)	Launch impact	Degradation of the optical quality due to the launch (e.g. vibrations move optics / detectors around)	4	10
		Space environment	Degradation of the optical quality due to the difference of environment between ground and space	5	10
		Thermal load	Varying thermal load will modify the optics and degrade the optical quality (e.g., shape of the M1 and M2 optics)	6	31
		Drift and jitter	Effect of the high-freq vibrations and LoS drift during integration	7	20.2
M1 deployable segments	Differential piston residual	Pist. Error between segments	Phasing residual between M1 segments, after active optics correction. Only tip, tilt, piston is controlled. Lateral X-Y movements are not, leading to additional WFE error.	8	20
	Differential tip-tilt residual	TT Error between segments	Orientation residual between M1 segments, after active optics correction.	9	20
		Focus error between M1 and M2	Compensation of focus error with M1 (i.e. individual segments have wrong RoC).	10	10
		M1 deployment precision	M1 deployment error	Only tip, tilt, piston is controlled. Lateral X-Y movements are not, leading to additional WFE error.	10.1
M2 deployable segments	Secondary mirror deployment precision	M2 deployment error	Only tip, tilt, piston is controlled. Lateral X-Y movements are not, leading to additional WFE error.	11	0
	Tip-tilt residual	M2 control	Residual M2 tip-tilt error (rX, rY)	12	15
	Focus residual	M2 control	Residual M2 focusing error (Z)	13	15
Contingency	Other unaccounted errors	Misc.		14	10
					72

STOP analysis



31 nm RMS

Active Optics

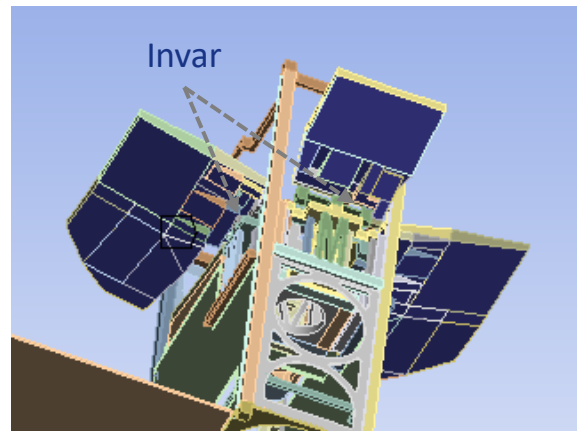
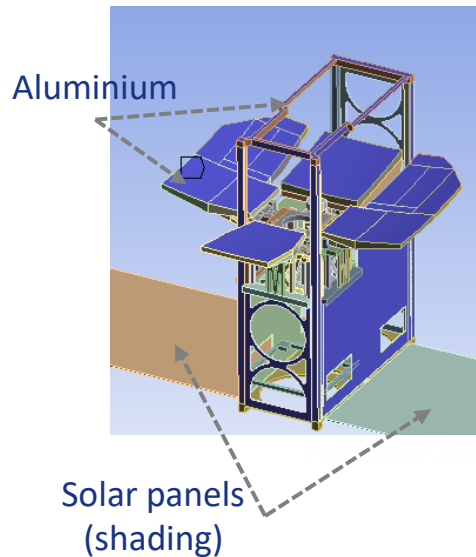


36 nm RMS

STOP analysis

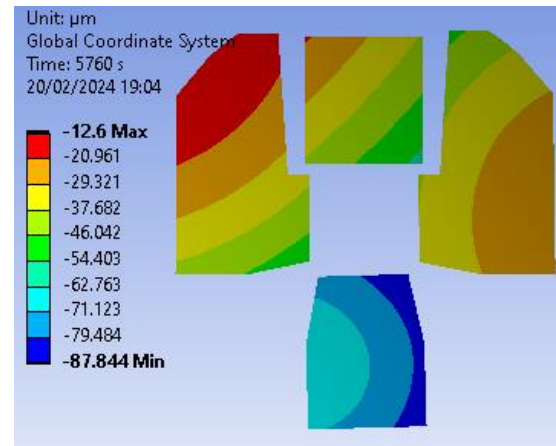
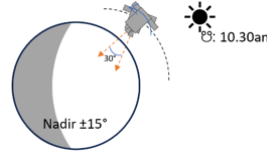
Structural-Thermal-Optical-Performance (STOP) Analysis

- The thermal environment changes the shape of the structure & optical elements
 - Aluminium frame and mirrors. Solar panels. Invar joints for segments
 - Sun radiation + Earth
- Modelling
 - ANSYS Workbench + OpticStudio

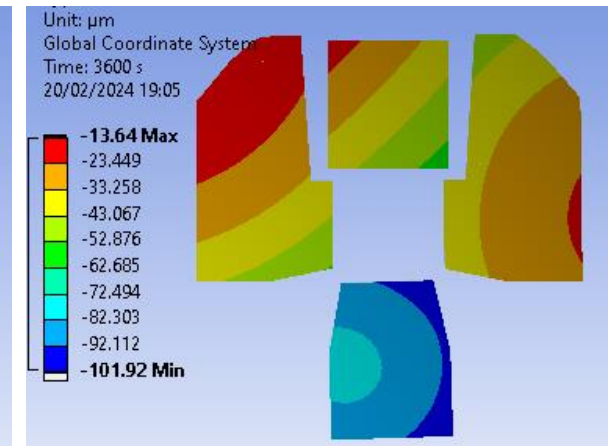


Preliminary results

- Several loading scenarios investigated
 - Sun synchronous orbit
 - Both for EO and astronomy scenarios
- Thermal deformations
 - Last instance in Sun/shadow are quite similar
 - Aberrations are mostly Piston, Tip-Tilt, Focus
 - Aberrations still high for visible wavelength
- Investigating mitigation strategies





In shadow



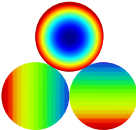

In Sun

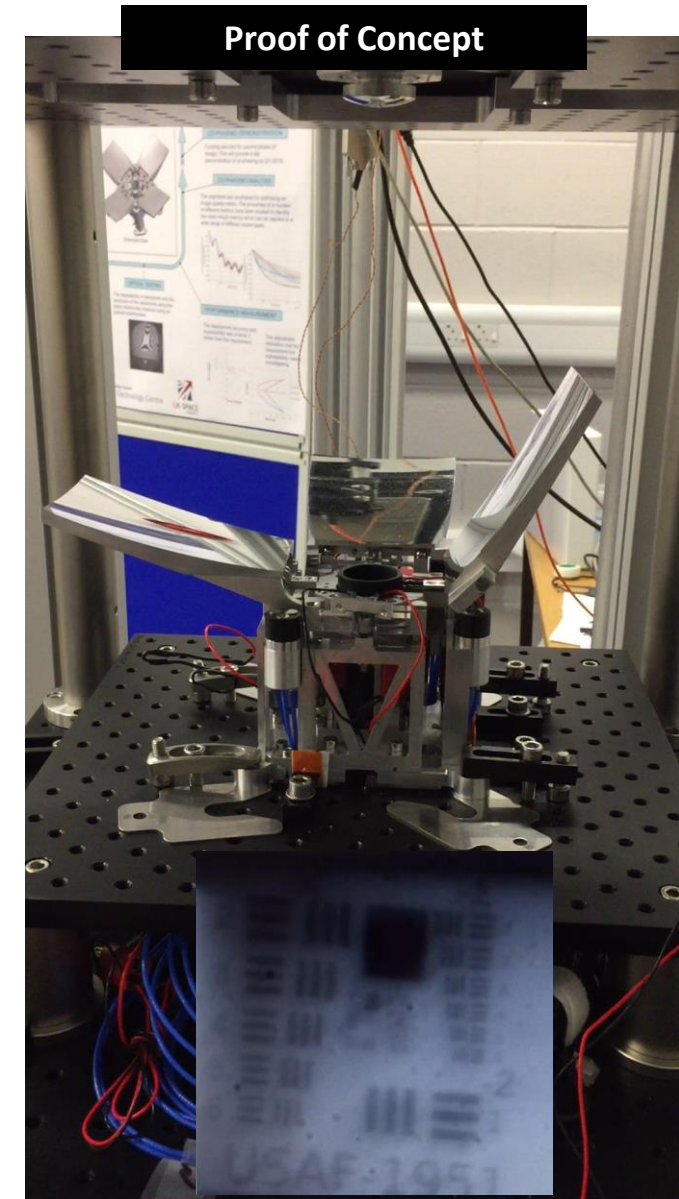
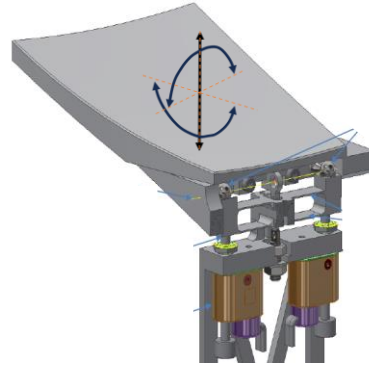
Active optics control

Align and phase telescope

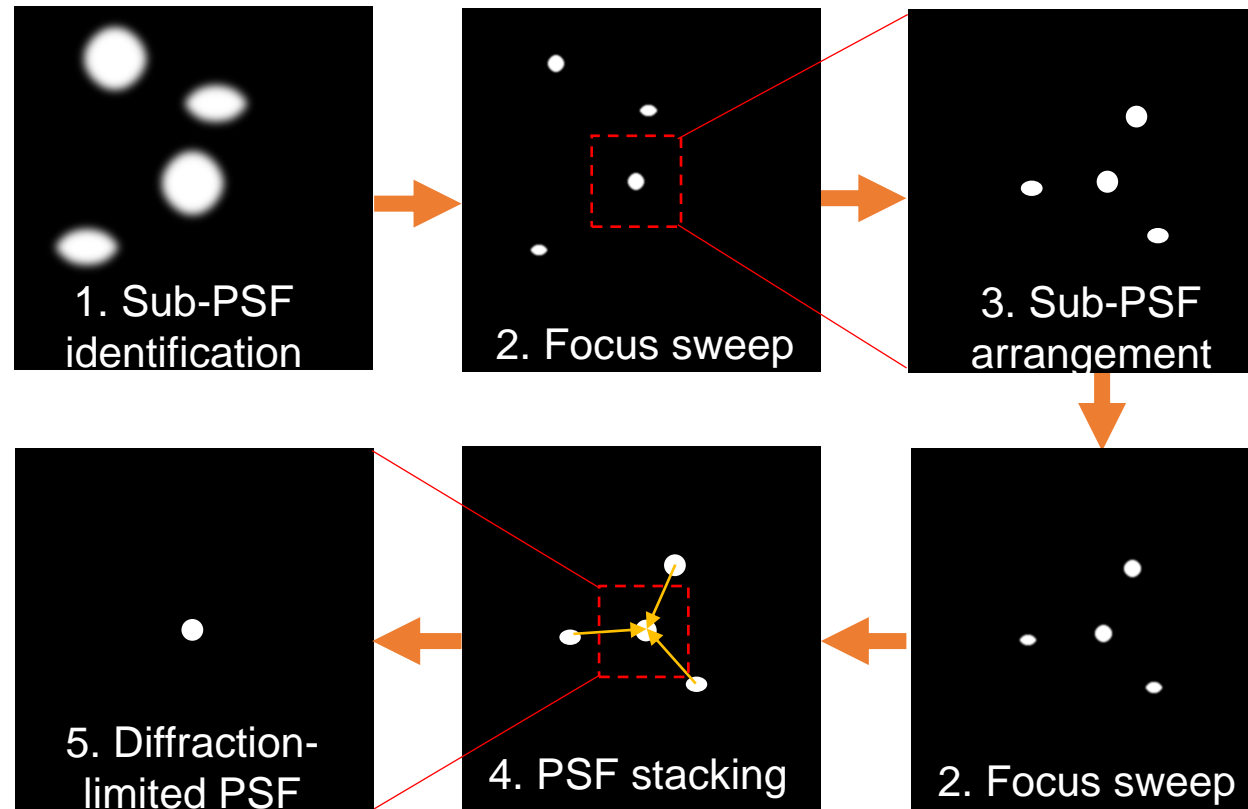
- After initial deployment 
- Maintain during observations 

Active optics

- Measurement
 - Displacement sensors
 - Image sensor
- Linear actuators
 - M1 & M2: correction tip, tilt, piston/focus 
- Control algorithm
 - Accurate, light (on-board), robust 
 - Work on point-source & extended objects

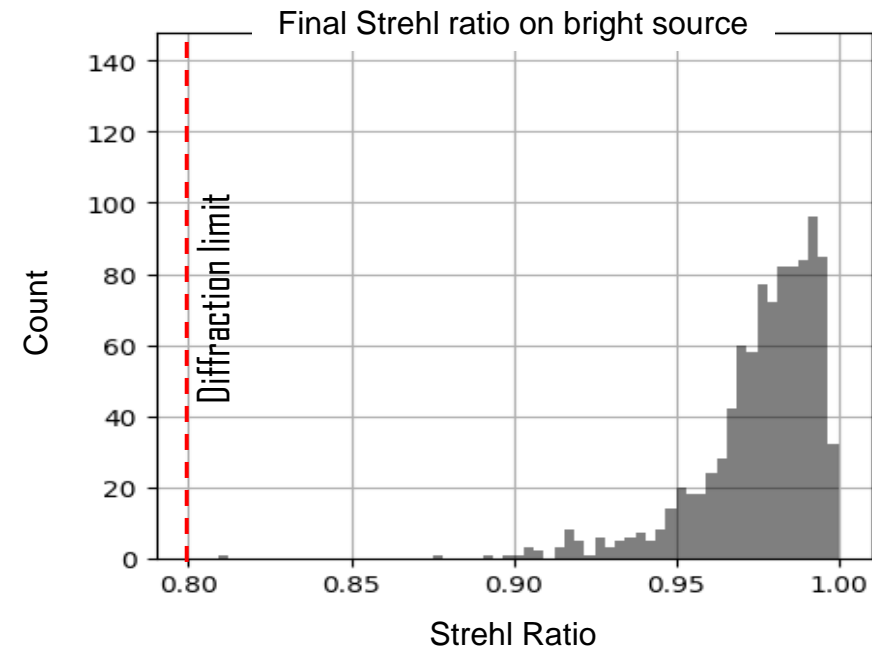
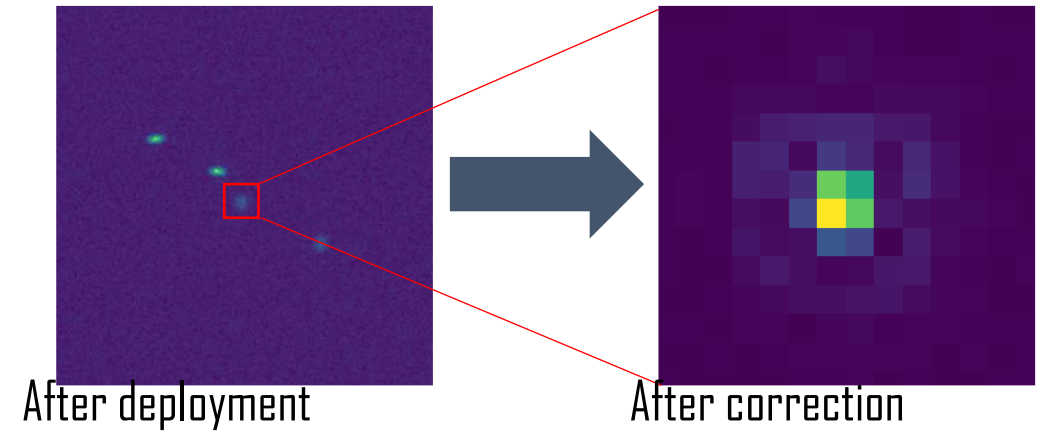


Classical approach (sequential)



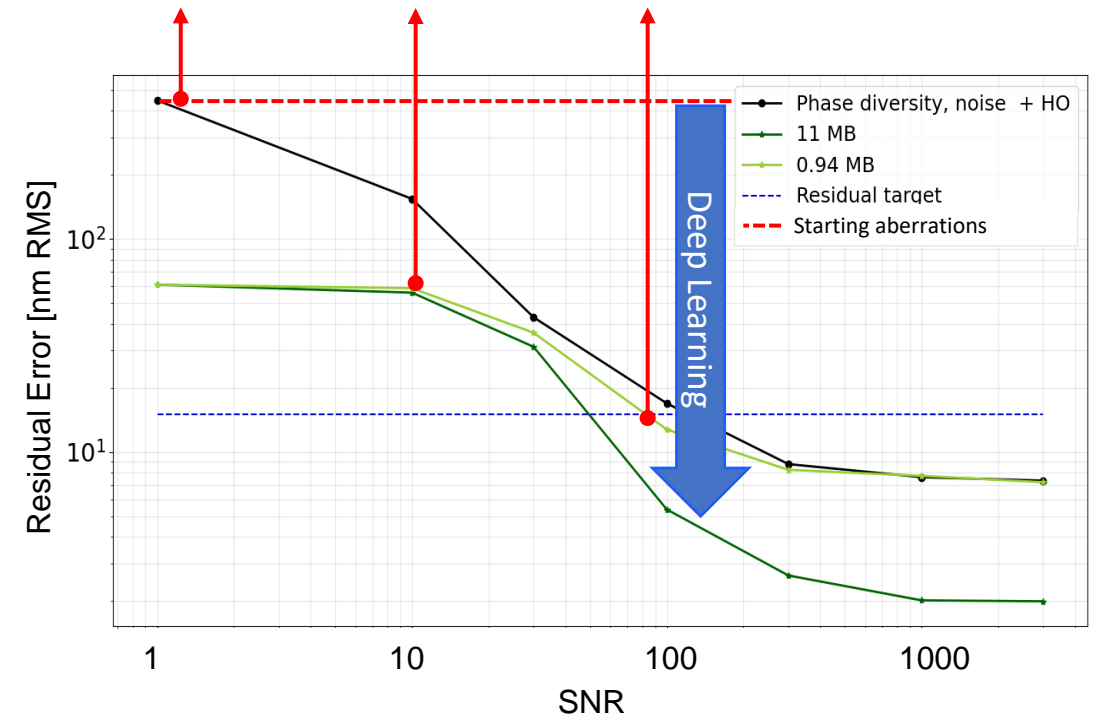
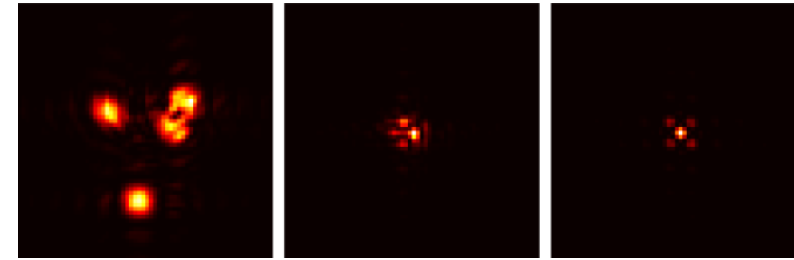
Classical approach (sequential)

- Sequential image sharpening
 - 40–50 images needed
 - $M = \sum_{x,y} I_{x,y}^2$
 - Low computational complexity
 - Robust, with error handling
- Excellent performance
 - SR > 85-90% for good SNRs (>60-70)
 - Working on improving performance for differential piston in low flux
 - Currently testing on optical bench



Deep Learning

- CCN adapted light ResNet architecture
 - 1st network for large tip-tilt
 - 2nd network for small tip-tilt & piston
- Only requires 2 images!
 - Reaches diffraction-limit
 - Small computational burden
 - Robust & stable



Maxime Dumont, et al, "Phasing segmented telescopes via deep learning methods: application to a deployable CubeSat," J. Opt. Soc. Am. A 41, 489-499 (2024)

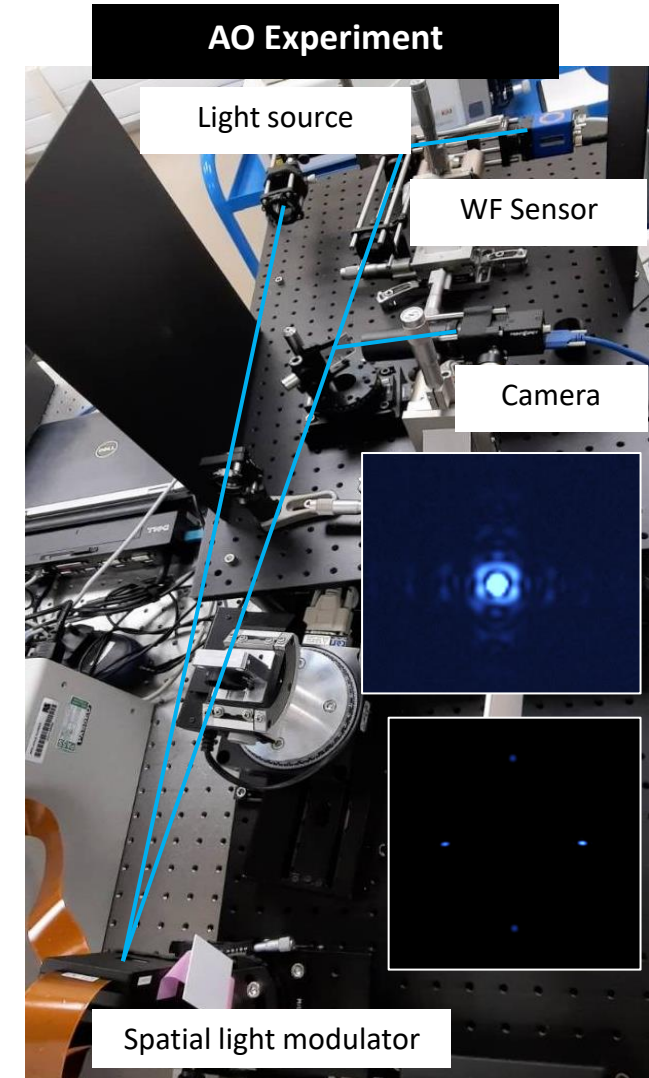
Conclusions & next steps

ADOT

- Development of new technology
- High-resolution in the optical
- Cost reduction over traditional platforms

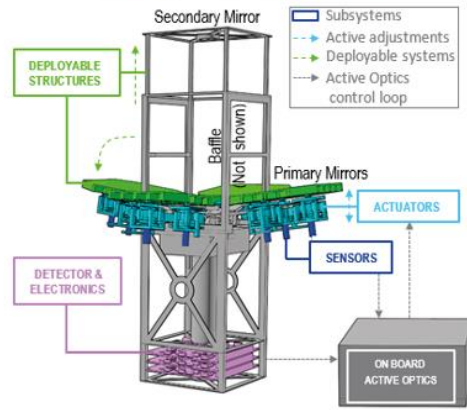
Next steps

- Adaptive optics
 - Laboratory validation
 - Improve differential piston correction
 - Fully extended objects
 - Data fusion (T°C and displacement)
- Optical-mechanical design
 - Improve thermal behaviour
 - Deployable M2



Thank you for your attention

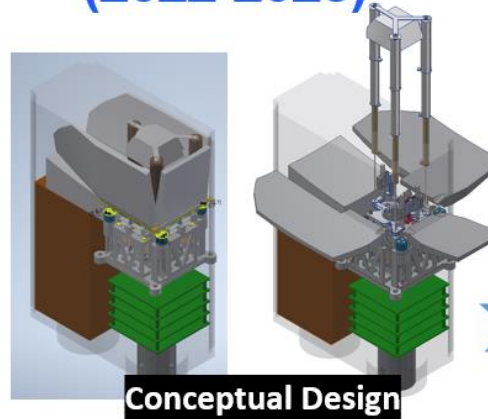
ISAAC - CLASP (2020-2022)



6U ground demonstration
 Satellite (payload + bus) concept
 Deployable M1 validation
 Full optical design
 Preliminary AO control

TRL3/4

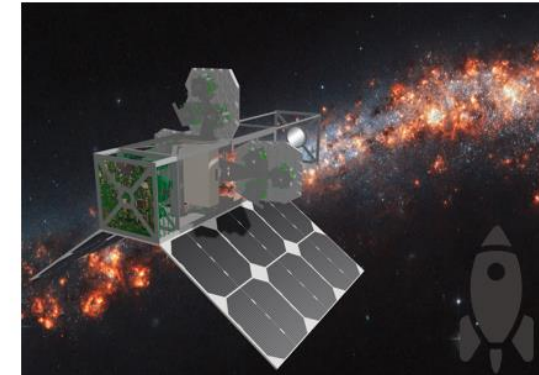
A-DOT (2022-2025)



6U ground demonstration
 Deployable M2 validation
 Preliminary electronics
 Full end-to-end AO control

TRL4

A-DOT+ (2025-2028)



Full spacecraft
 Payload prototype qualified in
 operational environment

TRL7